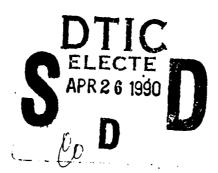


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# OXIDES OF NITROGEN EMISSIONS FROM THE TESTING OF TF41-A-2B ENGINES AT NAS LEMOORE, CALIFORNIA

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# OXIDES OF NITROGEN EMISSIONS FROM THE TESTING OF TF41-A-2B ENGINES AT NAS LEMOORE, CALIFORNIA

STATEMENT "A" per Douglas Marable Aircraft Environmental Support Office Naval Aviation Depot, Naval Air Station North Island, San Diego, CA 92135 IELECON 4/25/90 VG

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OXIDES OF NITROGEN EMISSIONS FROM THE TESTING OF TF41-A-2B ENGINES AT NAS LEMOORE, CALIFORNIA

AESO Report No. 1-88 November 1987

EXECUTIVE SUMMARY

Oxides of nitrogen  $(NO_X)$  are an air pollutant from the testing of gas turbine engines. Out-of-airframe engine testing is regulated by air pollution control agencies which require NOX emissions data on applications for permits to construct and operate engine test facilities. Aside from continuous emissions monitoring, current methods of determining NOx emissions from test cells depend on the availability of accurate records of engine operational data. This degree of record keeping is excessive given the difficult conditions under which engine testing is normally conducted. To avoid excessive record keeping, the Aircraft Environmental Support Office recommends a simple procedure for the determination of NOx' emissions. Its use depends only on accurate records of fuel usage for each engine test run.

The procedure involves the use of a correlation coefficient which relates the weight (pounds) of NOx emissions to the weight (pounds) of fuel consumed during engine testing. The coefficient is characteristic of a given engine type, demonstrating little variation among individual engines. This report establishes a correlation coefficient for the TF41-A-2B engine based on actual emissions data and the run sheets from 27 engine tests conducted in test cells at NAS Lemoore, California. The correlation coefficient, equal to 0.01515 pounds of NOx formed per pound of fuel consumed, determined NOT emissions to within 1% of actual values. An analysis of the statistical validity of the coefficient supports its use as a reliable procedure. Notice of the state of the fact of the state of the st

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OXIDES OF NITROGEN EMISSIONS FROM THE TESTING OF TF41-A-2B ENGINES AT NAS LEMOORE, CALIFORNIA

AESO Report No. 1-88 November 1987

#### 1. INTRODUCTION

Oxides of nitrogen  $(NO_X)$ , a combination primarily of nitrogen oxide and nitrogen dioxide, are an air pollutant from the testing of gas turbine engines. Engine test cells therefore constitute a stationary source of  $NO_X$  emissions. Unless otherwise exempt, newly constructed or modified test cells must comply with federal New Source Performance Standards. (1) These standards were designed to regulate operations with continuous emissions, such as power plants and refineries. However, engine test cells are inactive for many days of the year, and are therefore not continuous emission sources.

The amount of  $\mathrm{NO}_{\mathrm{X}}$  formed during engine testing is part of the information required on applications to regulatory agencies for authority to construct and permission to operate test facilities. To assist NAS Lemoore in complying with local emissions regulations, the Aircraft Environmental Support Office (AESO) developed a procedure for determining  $\mathrm{NO}_{\mathrm{X}}$  emissions. The procedure establishes a correlation coefficient which relates the weight of  $\mathrm{NO}_{\mathrm{X}}$  emissions produced to the weight of fuel consumed during an engine test run. The correlation coefficient can be used to determine  $\mathrm{NO}_{\mathrm{X}}$  emissions over a specified compliance period, based only on values of fuel usage per engine test run.

This report establishes a correlation coefficient for the TF41-A-2B engine based on a sample of 27 engine tests conducted in test cells at NAS Lemoore, California. An analysis of the statistical validity of the coefficient is presented to support its use. AESO used a similar approach in an earlier report to establish a correlation coefficient for the F404-GE-400 engine. (2) The general approach is described in the next section.

#### 2. EMISSION INDEXES AND ENGINE OPERATIONAL DATA

#### 2.1 EMISSION INDEXES

An emission index relates the amount of a pollutant in the engine exhaust to the amount of fuel used. It is commonly expressed as pounds of exhaust constituent per 1,000 pounds of fuel consumed. This report considers only a single pollutant, oxides of nitrogen. The emission index for NO  $_{\rm X}$  is calculated from measured concentrations of carbon monoxide, carbon dioxide, oxides of nitrogen and hydrocarbons in the engine exhaust.

The emission index for  $\mathrm{NO}_{\mathrm{X}}$  varies with engine power setting, being lowest at idle and highest at full power (military mode). In order to calculate the amount of  $\mathrm{NO}_{\mathrm{X}}$  formed at each power setting, an appropriate value of the emission index must be known. To establish these values, AESO evaluated the measured gaseous emissions data from 7 TF41-A-2B engine test reports. (3) Each test report provides values of the emission index for 6 power settings between idle and full power. Figure 1 shows a semi-logarithmic plot of emission index versus thrust for the reported gaseous emissions data. A simple curve fitting program was used to generate an exponential curve fit defined by the equation

$$y = ae^{bx}$$
 (1)

where a and b are constants (a = 2.02 and b =  $1.76 \times 10^{-4}$ ) determined by the program, y is the emission index and x is the thrust in pounds. The coefficient of determination,  $r^2$ , for the resulting fit is 0.93301.

Only data corresponding to thrust values in the range from 4,000 to 15,000 pounds were used to generate the curve fit. The TF41-A-2B engine is rated at 15,000 pounds of thrust. Therefore regular engine testing above this limit was not conducted. As seen from Figure 1, the relationship between emission index and thrust is not exponential for thrust values between idle and 4,000 pounds. For this reason an average value of 1.6 was calculated as the emission index for idle. Equation (1) is then used to calculate the emission index for thrust settings within the specified range, for any TF41-A-2B engine.

#### 2.2 ENGINE OPERATIONAL DATA

This section of the report describes the organization of the data in Tables 1 - 27. The TF41-A-2B engine run sheets on which these tables are based are included in Appendix A. Tables 1 - 9 contain engine data from test cells at NAS Lemoore. The tests were conducted from May through July of 1987. Tables 10 - 27 contain engine data from earlier tests also conducted in test cells at NAS Lemoore - testing occurred from August through November of 1985. The engine run sheets, on which these tables are based sometimes lacked thrust and fuel consumption values at idle and military operation. Estimated values are provided for these parameters. At idle the TF41-A-2B demonstrates an average thrust of 640 pounds and an average fuel flow rate of 1,050 pounds per hour. At military the average thrust is about 12,800 pounds, and the average fuel flow rate is 8,040 pounds per hour.

The first 4 columns of each table contain the engine operational parameters. These 4 parameters are: rpm, thrust, fuel flow, and time of operation at a specified power setting. For the purposes of this report, rpm is used only as a means of data identification. It is not used computationally. The rpm provides an indication of the relative power setting for each row of data in the tables, ranging from idle (7,000 rpm) to full power (12,800 rpm). Equation (1) and the remaining 3 parameters are then used to determine the entries in the next 3 columns; "Fuel use", "EI" and "Pounds of NO $_{\rm X}$ ". Section 2.3 describes the calculations and summarizes the results.

#### 2.3 CALCULATIONS AND SUMMARY OF RESULTS

#### 2.3.1 CALCULATIONS

Tables 1 - 27 present TF41-A-2B engine test data and the resulting NO $_{\rm X}$  emissions. The amount of NO $_{\rm X}$  formed at each power setting is determined as follows. Each row of data in the table, identified by an rpm, corresponds to a power setting. By using the fixed emission index at idle (1.60) and equation (1) for all other power settings, this report calculates an emission index for each recorded thrust value. The fuel usage (pounds) is calculated by multiplying the fuel flow by the time, which must first be converted into hours. A division of the fuel usage by 1,000 (the emission index is for pounds per 1,000 pounds of fuel used) and multiplication by the emission index then gives the "pounds of NO $_{\rm X}$ " formed at that power setting. A summation of the NO $_{\rm X}$  emissions from all the power settings in a table gives the total NO $_{\rm X}$  emissions for that engine test.

Tables 1 - 27 conclude by calculating the total fuel consumption and the total  $\mathrm{NO}_{\mathrm{X}}$  emissions for the test. A division of the total pounds of  $\mathrm{NO}_{\mathrm{X}}$  by the total pounds of fuel used gives the pounds of  $\mathrm{NO}_{\mathrm{X}}$  formed per pound of fuel. This value is the correlation coefficient discussed in the introduction. Note that this parameter has a function similar to that of the emission index, although it is expressed somewhat differently.

#### 2.3.2 COMPUTER PROGRAM

A FORTRAN program was used to perform the calculations and to generate the tables. The program documentation and coding appear in Appendix B. A catalog of environmental calculations, also available from AESO, describes how to use a small programmable calculator to establish emission indexes and to estimate the amounts of  $NO_x$  formed during engine testing.  $\{4\}$ 

#### 2.3.3 SUMMARY OF RESULTS

The results of Tables 1 - 27 are summarized in the first 1 columns of Table S1. Column 4 contains the correlation coefficient for the individual engine tests. The mean correlation coefficient for the entire sample is 0.01515 pounds of  $NO_{\rm x}$  formed per pound of fuel consumed. The standard deviation for the sample, equal to 0.00137, quantifies the variability of the individual test results.

Column 5 of the same table contains the calculated amounts of  $\rm NO_X$  emissions, which are derived from the mean correlation coefficient. The calculated amounts compare closely with the amounts determined from the emission indexes. Column 6 reports the percentage differences for the individual engine tests, which vary from -21.37% to +22.68%. However, the total calculated amount of  $\rm NO_X$  emissions for the sample agrees to within less than 1% of the actual amount. Although individual engine tests may demonstrate significant percentage differences, use of the correlation coefficient to determine  $\rm NO_X$  emissions produces reliable results for larger samples.

#### 3. STATISTICAL ANALYSIS

The sample used to establish the correlation coefficient in the previous section is only a limited subset of all TF41-A-2B engine tests. However, it is desirable to infer from this sample about characteristics pertaining to the entire population. Statistical inference from a sample is useful only if that sample has been randomly chosen, and is representative of the population from which it was obtained. The 27 TF41-A-2B engine tests evaluated in this report were chosen because they were conducted during 2 arbitrarily established calendar intervals, one in 1985 and the other in 1987. The selection was therefore random, and the sample should be considered representative of the population.

To demonstrate that the use of the correlation coefficient is a statistically valid means of determining NO $_{\rm X}$  emissions, it is necessary to show that any apparent discrepancies between the actual and calculated amounts of NO $_{\rm X}$  are due only to random sampling error, and not to failure of the procedure. The 2 parameters of interest, actual and calculated NO $_{\rm X}$  emissions, are not independent of each other; they are paired through the procedure used to establish the correlation coefficient. The applicable statistical method is the t-test for paired observations, which is described in many standard texts on probability and statistics. (5) An extension of the t-test for paired observations is recognized by the Environmental Protection Agency as a method for certifying NO $_{\rm X}$  monitoring equipment. (6)

Table S2 presents the results of the statistical analysis, and provides a brief outline of the equations used in the calculation procedure. The first 3 columns of the table duplicate information from Table S1, while the next 2 columns report the intermediate results. The important statistical parameters are then summarized at the bottom of the table.

The values in column 4 are the differences between the actual and the calculated amounts of  $\mathrm{NO}_{\mathrm{X}}$  for each engine test. The estimated standard deviation, a value which describes the variability among these differences, is used to derive a standard error for the mean of differences. The standard error quantifies how well the mean difference for the sample estimates the mean difference for the population, which is usually zero. From the estimated standard error, and an appropriate t-value, confidence limits for the sample mean difference can be determined.

95% confidence limits, based on a two-tailed test, are appropriate. The expected confidence interval for the sample mean difference is then 1.66  $\pm$  5.40. The absolute magnitude of this confidence interval determines the relative accuracy, which describes how closely the proposed procedure compares to the reference method (calculation of NO\_x emissions based on the emission indexes). The relative accuracy determined from this statistical analysis is 3.79%. This means that one can be 95% confident that the correlation coefficient will determine NO\_x emissions which are within 3.79% of emissions calculated by using the emission indexes.

#### 4. CONCLUSIONS

The rate at which the testing of TF41-A-2B engines produces  $\mathrm{NO}_{\mathrm{X}}$  emissions can be expressed as a correlation coefficient. This coefficient can then be used to determine the amount of  $\mathrm{NO}_{\mathrm{X}}$  emissions formed during any engine test, based only on records of fuel consumption. The correlation coefficient for the TF41-A-2B engine is 0.01515 pounds of  $\mathrm{NO}_{\mathrm{X}}$  formed per pound of fuel consumed. An analysis of the statistical validity of the coefficient supports its use.

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- 1. Code of Federal Regulations, Title 40, Part 60, "Subpart GG Standards of Performance for Stationary Gas Turbines," Washington DC, July, 1985.
- 2. Aircraft Environmental Support Office, "Emissions of Nitrogen Oxides from the Testing of F404-GE-400 Engines at Naval Air Station, Lemoore, California," AESO Report No. 4-85, July, 1985.
- 3. Scott Environmental Technology, Inc., "Individual Engine Test & Model Summary Reports, Modification 6, Alameda Testing, USAF Contract No. F29601-75-C-0046," Prepared for: Air Force Civil Engineering Center, Tyndall Air Force Base, Florida, October 20, 1976.
- 4. Aircraft Environmental Support Office, "Environmental Calculations: A Handbook for the Use of Small Programmable Calculators to Determine the Effect of Aircraft on the Environment," AESO Report No. 8-86, May, 1987.
- 5. Klugh, Henry E., "Statistics: The Essentials for Research," 2nd ed., John Wiley & Sons, Inc., New York, 1974.
- 6. Code of Federal Regulations, Title 40, Part 60, "Appendix B Performance Specification 2 Specifications and Test Procedures for  ${\rm SO}_2$  and  ${\rm NO}_{\rm X}$  Continuous Emission Monitoring Systems in Stationary Sources," Washington DC, July, 1985.

TABLE S1. Summary of oxides of nitrogen emissions from the testing of TF41-A-2B engines at NAS Lemoore

Table	Pounds of NOx	Pounds of	Pounds of NOx	Pounds of NOx	% difference
	per test	fuel used	per pound of	(calculated)	
		in test	fuel used in test	(a) <sup>.</sup>	(b)
			test		
1	72.06	5099.4	0.01413	77.26	7.22
2	54.38	4087.2	0.01330	61.92	13.87
3	214.77	14579.0	0.01473	220.87	2.84
4	111.37	7038.3	0.01582	106.63	-4.26
5	42.03	2743.3	0.01532	41.56	-1.12
6	122.26	8096.9	0.01510	122.67	0.34
7	111.45	7739.3	0.01440	117.25	5.20
8	114.96	8019.2	0.01434	121.49	5.68
9	185.04	13081.0	0.01415	198.18	7.10
10	192.06	14427.7	0.01331	218.58	13.81
11	524.00	33635.1	0.01558	509.57	<b>-2.7</b> 5
12	473.31	30676.1	0.01543	464.74	-1.81
13	338.98	22433.6	0.01511	339.87	0.26
14	140.86	9630.8	0.01463	145.91	3.59
15	182.55	9474.5	0.01927	143.54	-21.37
16	165.47	10128.4	0.01634	153.45	<del>-</del> 7.26
17	168.64	12132.5	0.01390	183.81	9.00
18	202.57	12454.8	0.01626	188.69	-6.85
19	122.77	7748.0	0.01584	117.38	<del>-</del> 4.39
20	60.24	4878.1	0.01235	73.90	22.68
21	108.22	7446.8	0.01453	112.82	4.25
22	169.24	11738.5	0.01442	177.84	5.08
23	222.60	14510.3	0.01534	219.83	-1.24
24	211.86	12540.3	0.01689	189.99	-10.32
25	330.69	21292.3	0.01553	322.58	<del>-</del> 2.45
26	173.44	10563.1	0.01642	160.03	<b>-7.7</b> 3
27	212.38	12734.2	0.01668	192.92	<del>-</del> 9.16
	<del></del>				
totals	5 5028.20			4983.28	-0.89
		mean	0.01515		
		7 2 - 4 - 2	0.01313		•

mean 0.01515 standard deviation 0.00137

<sup>(</sup>a) Pounds of NOx (calculated) is determined by multiplying the pounds of fuel used in a given test by the mean correlation coefficient as established from the 27 actual test runs.

<sup>(</sup>b) % difference = pounds of NOx (calc'd) - pounds of NOx (per test) x 100 pounds of NOx (per test)

TABLE 1. Emission of oxides of nitrogen from the testing of a TF41 engine at NAS Lemoore (Engine Serial Number: 141011)

Date:	7/ 7/198	37					
RPM	THRUST (lb)	FUEL FLOW (lb/hr)	TIME (MIN)	FUEL USE (lb)	EI	POUNDS NOX	
12001	8810	5578	6	557 <b>.</b> 8	9.52	5.31	
12187	9920	6346	6	634.6	11.58	7.35	
12427	11380	7462	5	621.8	14.97	9.31	
12645	12440	8343	6	834.3	18.04	15.05	
11958	8420	5351	5	445.9	8.89	3.96	
12230	10010	6483	5	540.2	11.76	6.35	
12480	11460	7562	5	630.2	15.18	9.57	
12694	12480	8345	6	834.5	18.17	15.16	
				<del></del>		<del></del>	
Pounds of fuel used in test 5099.4							
Pounds of NOx per test 72.06							

<sup>(</sup>a) An rpm of 7000 indicates that the engine is operating at idle. The corresponding thrust and fuel flow are estimated.

<sup>(</sup>b) An rpm of 12800 indicates that the engine is operating at military. The corresponding thrust and fuel flow are estimated.

TABLE 2. Emission of oxides of nitrogen from the testing of a TF41 engine at NAS Lemoore (Engine Serial Number: 141505)

Date:	7/ 9/198	37				
RPM	THRUST (lb)	FUEL FLOW (lb/hr)	TIME (MIN)	FUEL USE (lb)	EI	POUNDS NOX
11991	8550	5525	5	460.4	9.10	4.19
12192	9950	6410	5	534.2	11.64	6.22
12531	11480	7614	5	634.5	15.23	9.67
12786	12520	8557	5	713.1	18.29	13.05
12018	8550	5625	6	562.5	9.10	5.12
12225	10040	6525	5	543.7	11.82	6.43
12543	11470	7665	5	638.7	15.21	9.71
						***************************************
Pounds	of fuel	used in test		4087.2		
Pounds	of NOx p	er test				54.38

<sup>(</sup>a) An rpm of 7000 indicates that the engine is operating at idle. The corresponding thrust and fuel flow are estimated.

<sup>(</sup>b) An rpm of 12800 indicates that the engine is operating at military. The corresponding thrust and fuel flow are estimated.

TABLE 3. Emission of oxides of nitrogen from the testing of a TF41 engine at NAS Lemoore (Engine Serial Number: 141430)

Date: 6/23/1987

RPM	THRUST (1b)	FUEL FLOW (lb/hr)	TIME (MIN)	FUEL USE (lb)	EI	POUNDS NOX
11712	8570	5281	5	440.1	9.13	4.02
11969	10010	6243	5	520.2	11.76	6.12
12334	11540	7280	5	606.7	15.40	9.34
12815	13450	9040	5	753.3	21.55	16.23
11739	8430	5246	5	437.2	8.91	3.89
12030	10060	6363	5	530.2	11.87	6.29
12308	11420	7375	5	614.6	15.07	9.26
12684	13310	8793	5	732.7	21.02	15.41
11895	8500	5381	6	538.1	9.02	4.85
12144	9970	6363	7	742.3	11.68	8.67
12435	11380	7415	7	865.1	14.97	12.95
12780	12960	8758	5	729.8	19.77	14.43
11954	8500	5409	8	721.2	9.02	6.50
12208	10020	6464	7	754.1	11.78	8.89
12472	11350	7418	8	989.1	14.89	14.73
12779	12880	8633	7	1007.2	19.49	19.63

<sup>(</sup>a) An rpm of 7000 indicates that the engine is operating at idle. The corresponding thrust and fuel flow are estimated.

<sup>(</sup>b) An rpm of 12800 indicates that the engine is operating at military. The corresponding thrust and fuel flow are estimated.

TABLE 3 (continued)

RPM	THRUST (lb)	FUEL FLOW (lb/hr)	TIME (MIN)	FUEL USE (lb)	EI	POUNDS NOX
12016	8680	5545	6	554.5	9.31	5.16
12248	10160	6553	8	8,73.7	12.08	10.55
12522	11470	7550	7	880.8	15.21	13.40
12788	12730	8586	9	1287.9	18.98	24.45
Pounds of fuel used in test				14579.0		
Pounds	of NOx pe	er test				214.77

<sup>(</sup>a) An rpm of 7000 indicates that the engine is operating at idle. The corresponding thrust and fuel flow are estimated.

<sup>(</sup>b) An rpm of 12800 indicates that the engine is operating at military. The corresponding thrust and fuel flow are estimated.

TABLE 4. Emission of oxides of nitrogen from the testing of a TF41 engine at NAS Lemoore (Engine Serial Number: 141011)

Date:	5/21/198	37				
RPM	THRUST (1b)	FUEL FLOW (lb/hr)	TIME (MIN)	FUEL USE (lb)	EI	POUNDS NOX
11743	8630	5959	8	794.5	9.23	7.33
11979	10130	6639	8	885.2	12.01	10.63
12218	11500	7723	7	901.0	15.29	13.78
12596	13270	8847	9	1327.1	20.88	27.70
11738	8550	5356	7	624.9	9.10	5.68
12022	10170	6449	6	644.9	12.10	7.80
12274	11580	7475	6	747.5	15.51	11.59
12811	14090	9542	7	1113.2	24.12	26.85
Pounds	of fuel	used in test		7038.3		
Pounds of NOx per test 111.37						

<sup>(</sup>a) An rpm of 7000 indicates that the engine is operating at idle. The corresponding thrust and fuel flow are estimated.

<sup>(</sup>b) An rpm of 12800 indicates that the engine is operating at military. The corresponding thrust and fuel flow are estimated.

TABLE 5. Emission of oxides of nitrogen from the testing of a TF41 engine at NAS Lemoore (Engine Serial Number: 142542)

Date:	6/16/198	37				
RPM	THRUST (lb)	FUEL FLOW (lb/hr)	TIME (MIN)	FUEL USE (lb)	EI	POUNDS NOX
11804	8430	5320	5	443.3	8.91	3.95
12124	10320	6586	6	658.6	12.42	8.18
12364	11500	7483	5	623.6	15.29	9.53
12686	13030	8724	7	1017.8	20.01	20.37
Pounds	of fuel	used in test		2743.3		
Pounds of NOx per test						42.03

<sup>(</sup>a) An rpm of 7000 indicates that the engine is operating at idle. The corresponding thrust and fuel flow are estimated.

<sup>(</sup>b) An rpm of 12800 indicates that the engine is operating at military. The corresponding thrust and fuel flow are estimated.

TABLE 6. Emission of oxides of nitrogen from the testing of a TF41 engine at NAS Lemoore (Engine Serial Number: 141483)

Date:	6/18/198	37				
RPM	THRUST (lb)	FUEL FLOW (lb/hr)	TIME (MIN)	FUEL USE (lb)	EI	POUNDS NOX
11707	8760	5479	5	456.6	9.44	4.31
11993	10160	6398	5	533.2	12.08	6.44
12258	11410	7313	6	731.3	15.05	11.00
12651	13210	8852	5	737.7	20.66	15.24
11707	8470	5310	6	531.0	8.97	4.76
12028	10020	6381	7	744.5	11.78	8.77
12319	11340	7357	6	735.7	14.86	10.94
12742	13380	8962	6	896.2	21.28	19.08
11772	8630	5530	6	553.0	9.23	5.10
12055	10040	6371	5	530.9	11.82	6.28
12344	11510	7441	6	744.1	15.32	11.40
12767	13300	9028	6	902.8	20.99	18.95
Pounds	of fuel	used in test	:	8096.9		
Pounds	of NOx 1	per test				122.26

<sup>(</sup>a) An rpm of 7000 indicates that the engine is operating at idle. The corresponding thrust and fuel flow are estimated.

<sup>(</sup>b) An rpm of 12800 indicates that the engine is operating at military. The corresponding thrust and fuel flow are estimated.

TABLE 7. Emission of oxides of nitrogen from the testing of a TF41 engine at NAS Lemoore (Engine Serial Number: 141931)

Date:	6/24/198	37				
RPM	THRUST (1b)	FUEL FLOW (lb/hr)	TIME (MIN)	FUEL USE (lb)	EI	POUNDS NOX
11905	8670	5651	5	470.9	9.29	4.38
12097	10010	6426	5	535.5	11.76	6.30
12348	11410	7389	6	738.9	15.05	11.12
12697	13050	8827	5	735.6	20.08	14.77
11945	8580	5555	5	462.9	9.14	4.23
12185	10020	6608	5	550.7	11.78	6.49
12461	11520	7537	6	753.7	15.34	11.56
12620	12300	8200	6	820.0	17.60	14.43
11925	8570	5459	7	636.9	9.13	5.81
12177	10070	6551	5	545.9	11.89	6.49
12438	11550	7668	5	639.0	15.42	9.86
12644	12690	8493	6	849.3	18.85	16.01
Pounds of fuel used in test				7739.3		
Pounds of NOx per test 111.45						

<sup>(</sup>a) An rpm of 7000 indicates that the engine is operating at idle. The corresponding thrust and fuel flow are estimated.

<sup>(</sup>b) An rpm of 12800 indicates that the engine is operating at military. The corresponding thrust and fuel flow are estimated.

TABL 8. Emission of oxides of nitrogen from the testing of a TF41 engine at NAS Lemoore (Engine Serial Number: 141619)

Date:	7/10/198	37				
RPM	THRUST (lb)	FUEL FLOW (lb/hr)	TIME (MIN)	FUEL USE (lb)	EI	POUNDS NOX
11945	8740	5630	5	469.2	9.41	4.41
12160	10000	6500	6	650.0	11.74	7.63
12467	11460	7561	6	756.1	15.18	11.48
12618	12220	8165	6	816.5	17.35	14.17
11931	8660	5491	6	549.1	9.27	5.09
12108	10000	6520	5	543.3	11.74	6.38
12469	11520	7589	7	885.4	15.34	13.58
12712	12570	8486	6	848.6	18.46	15.66
11945	8520	5465	5	455.4	9.05	4.12
12239	10180	6663	5	555.2	12.12	6.73
12506	11590	7653	5	637.7	15.53	9.91
12733	12590	8526	6	852.6	18.52	15.79
Pounds	of fuel	used in test		8019.2		
Pounds of NOx per test 114.96						

<sup>(</sup>a) An rpm of 7000 indicates that the engine is operating at idle. The corresponding thrust and fuel flow are estimated.

<sup>(</sup>b) An rpm of 12800 indicates that the engine is operating at military. The corresponding thrust and fuel flow are estimated.

TABLE 9. Emission of oxides of nitrogen from the testing of a TF41 engine at NAS Lemoore (Engine Serial Number: 141359)

Date:	7/15/198	37				
RPM	THRUST (lb)	FUEL FLOW (lb/hr)	TIME (MIN)	FUEL USE (lb)	EI	POUNDS NOX
11849	8450	5271	7	615.0	8.94	5.50
12183	10040	6390	7	745.5	11.82	8.81
12490	11350	7410	6	741.0	14.89	11.03
12708	12350	8230	8	1097.3	17.76	19.48
11930	8500	5400	6	540.0	9.02	4.87
12165	9990	6375	7	743.7	11.72	8.72
12485	11470	7485	7	873.2	15.21	13.28
12646	12100	8020	7	935.7	16.99	15.90
11975	8550	5440	7	634.7	9.10	5.77
12230	10020	6503	7	758.7	11.78	8.94
12525	11450	7545	8	1006.0	15.15	15.25
12760	12550	8435	7	984.1	18.39	18.10
11985	8540	5430	7	633.5	9.08	5.75
12250	9950	6420	6	642.0	11.64	7.47
12545	11450	7515	8	1002.0	15.15	15.18
12790	12610	8465	8	1128.7	18.59	20.98
Pounds	s of fuel	used in tes	t	13081.0		
Pounds of NOx per test						185.04

<sup>(</sup>a) An rpm of 7000 indicates that the engine is operating at idle. The corresponding thrust and fuel flow are estimated.

<sup>(</sup>b) An rpm of 12800 indicates that the engine is operating at military. The corresponding thrust and fuel flow are estimated.

TABLE 10. Emission of oxides of nitrogen from the testing of a TF41 engine at NAS Lemoore (Engine Serial Number: 141252)

Date:	8/ 2/198	35				
RPM	THRUST (lb)	FUEL FLOW (lb/hr)	TIME (MIN)	FUEL USE (lb)	EI	POUNDS NOX
7000 <sup>a</sup>	640	1050	150	2625.0	2.26	5.93
11450	8762	5478	5	456.5	9.44	4.31
11798	10585	6762	5	563.5	13.01	7.33
12009	11826	7531	12	1506.2	16.19	24.39
12272	13091	8553	5	712.7	20.23	14.42
11318	8304	5152.	5	429.3	8.71	3.74
11492	9320	5814	5	484.5	10.42	5.05
11763	10741	6791	10	1131.8	13.38	15.14
12116	12612	8161	7	952.1	18.59	17.70
11355	8572	5320	6	532.0	9.13	4.86
11643	10201	6421	6	642.1	12.16	7.81
11972	11865	7634	6	763.4	16.30	12.45
12101 b	12545	8130	7	948.5	18.38	17.43
12800	12800	8040	20	2680.0	19.22	51.51
Pounds	of fuel	used in test		14427.7		
Pounds of NOx per test 192.						

<sup>(</sup>a) An rpm of 7000 indicates that the engine is operating at idle. The corresponding thrust and fuel flow are estimated.

<sup>(</sup>b) An rpm of 12800 indicates that the engine is operating at military. The corresponding thrust and fuel flow are estimated.

TABLE 11. Emission of oxides of nitrogen from the testing of a TF41 engine at NAS Lemoore (Engine Serial Number: 141298)

8/12/1985 Date: POUNDS NOX TIME FUEL USE EI FUEL FLOW THRUST RPM (lb) (MIN) (lb/hr) (lb) а 3850.0 2.26 8.70 220 1050 7000 640 737.5 9.44 6.96 8 8760 5531 11449 9 977.8 12.59 12.31 10396 6519 11726 1703.1 15.13 25.77 14 7299 11941 11442 18.79 7 980.2 19.17 12786 8402 12221 9.77 6.38 7 653.2 5599 8953 11476 9.36 12.38 7 756.3 6483 10301 11725 1005.5 16.14 16.23 8 7541 12012 11809 26.24 1187.3 22.10 8 8905 13594 12346 5.06 6 542.6 9.33 5426 8691 11405 9.72 6 701.0 13.87 7010 11849 10948 16.23 12.36 6 761.6 7616 11841 12020 968.8 25.52 24.73 9688 6 14412 12520 9.67 5.37 6 555.3 5553 11446 8895 7.87 6 641.9 12.26 6419 11694 10247

Emission index (EI) for NOx is expressed as pounds of NO  $_{\rm 2}$  per 1000 pounds of fuel.

12

7646

12023

11851

1529.2

16.26

24.87

<sup>(</sup>a) An rpm of 7000 indicates that the engine is operating at idle. The corresponding thrust and fuel flow are estimated.

<sup>(</sup>b) An rpm of 12800 indicates that the engine is operating at military. The corresponding thrust and fuel flow are estimated.

TABLE 11 (continued)

RPM	THRUST (lb)	FUEL FLOW (lb/hr)	TIME (MIN)	FUEL USE (lb)	EI	POUNDS NOX
12219	12922	8425	8	1123.3	19.64	22.06
11422	8699	5391	6	539.1	9.34	5.03
11693	10256	6423	6	642.3	12.28	7.89
12010	11824	7565	6	756.5	16.19	12.24
12516	14339	9624	6	962.4	25.20	24.25
12800 b	12800	8040	90	12060.0	19.22	231.78
				<del></del>		
Pounds of fuel used in test				33635.1		
Pounds	524.00					

<sup>(</sup>a) An rpm of 7000 indicates that the engine is operating at idle. The corresponding thrust and fuel flow are estimated.

<sup>(</sup>b) An rpm of 12800 indicates that the engine is operating at military. The corresponding thrust and fuel flow are estimated.

TABLE 12. Emission of oxides of nitrogen from the testing of a TF41 engine at NAS Lemoore (Engine Serial Number: 141234)

Date: 10/5/1985

RPM	THRUST (lb)	FUEL FLOW (lb/hr)	TIME (MIN)	FUEL USE (lb)	EI	POUNDS NOX
7000 a	640	1050	140	2450.0	2.26	5.54
11479	8767	5338	8	711.7	9.45	6.73
11765	10346	6375	8	850.0	12.48	10.61
12069	11910	7511	14	1752.6	16.43	28.80
11589	9278	5727	8	763.6	10.34	7.90
11901	10776	6870	8	916.0	13.46	12.33
12203	12449	8039	7	937.9	18.07	16.95
12568	14234	9506	9	1425.9	24.74	35.27
11518	8894	5426	8	723.5	9.66	6.99
11757	10248	6358	8	847.7	12.26	10.40
12020	11598	7336	12	1467.2	15.55	22.82
12350	13216	8599	8	1146.5	20.68	23.71
11539	8964	5495	8	732.7	9.78	7.17
11787	10465	6411	8	854.8	12.74	10.89
12017	11644	7285	8	971.3	15.68	15.23
12425	13563	8877	8	1183.6	21.98	26.02

<sup>(</sup>a) An rpm of 7000 indicates that the engine is operating at idle. The corresponding thrust and fuel flow are estimated.

<sup>(</sup>b) An rpm of 12800 indicates that the engine is operating at military. The corresponding thrust and fuel flow are estimated.

TABLE 12 (continued)

RPM	THRUST (1b)	FUEL FLOW (lb/hr)	TIME (MIN)	FUEL USE (lb)	EI	POUNDS NOX
11610	8784	5532	5	461.0	9.48	4.37
11843	10289	6499	5	541.6	12.35	6.69
12156	11770	7665	11	1405.2	16.03	22.53
11538	8603	5400	5	450.0	9.18	4.13
11800	10230	6483	5	540.2	12.23	6.61
12113	11940	7779	10	1296.5	16.52	21.42
12371	13356	8765	6	876.5	21.19	18.58
12800 b	12800	8040	55	7370.0	19.22	141.64
						<del></del>
Pounds	of fuel v	used in test		30676.1		
Pounds	473.31					

<sup>(</sup>a) An rpm of 7000 indicates that the engine is operating at idle. The corresponding thrust and fuel flow are estimated.

<sup>(</sup>b) An rpm of 12800 indicates that the engine is operating at military. The corresponding thrust and fuel flow are estimated.

TABLE 13. Emission of oxides of nitrogen from the testing of a TF41 engine at NAS Lemoore (Engine Serial Number: 141494)

Date: 9/10/1985

RPM	THRUST (lb)	FUEL FLOW (lb/hr)	TIME (MIN)	FUEL USE (lb)	EI	POUNDS NOX
7000 a	640	1050	175	3062.5	2.26	6.92
11617	8958	5455	5	454.6	9.77	4.44
11844	10264	6438	5	536.5	12.30	6.60
12089	11850	7383	9	1107.4	16.26	18.01
12541	14120	9281	6	928.1	24.25	22.50
11603	8901	5442	6	544.2	9.68	5.27
11833	10459	6492	6	649.2	12.73	8.26
12083	11848	7408	7	864.3	16.25	14.05
12522	14047	9225	7	1076.2	23.94	25.76
11613	8957	5521	6	552.1	9.77	5.40
11883	10751	6685	7	779.9	13.40	10.45
12101	11983	7570	7	883.2	16.64	14.70
11588	8971	5484	5	457.0	9.80	4.48
11766	10298	6318	5	526.5	12.37	6.51
12042	11869	7494	5	624.5	16.31	10.19
12386	13573	8812	5	734.3	22.02	16.17

<sup>(</sup>a) An rpm of 7000 indicates that the engine is operating at idle. The corresponding thrust and fuel flow are estimated.

<sup>(</sup>b) An rpm of 12800 indicates that the engine is operating at military. The corresponding thrust and fuel flow are estimated.

TABLE 13 (continued)

RPM	THRUST (lb)	FUEL FLOW (lb/hr)	TIME (MIN)	FUEL USE (lb)	EI	POUNDS NOX
11536	8747	5317	5	443.1	9.42	4.17
11786	10309	6408	5	534.0	12.40	6.62
12067	11887	7494	6	749.4	16.37	12.26
12429	13738	8965	6	896.5	22.67	20.32
ь 12800	12800	8040	45	6030.0	19.22	115.89
Pounds	of fuel	used in test		22433.6		
Pounds	338.98					

<sup>(</sup>a) An rpm of 7000 indicates that the engine is operating at idle. The corresponding thrust and fuel flow are estimated.

<sup>(</sup>b) An rpm of 12800 indicates that the engine is operating at military. The corresponding thrust and fuel flow are estimated.

TABLE 14. Emission of oxides of nitrogen from the testing of a TF41 engine at NAS Lemoore (Engine Serial Number: 141477)

Date:	8/26/198	35				
RPM	THRUST (lb)	FUEL FLOW (lb/hr)	TIME (MIN)	FUEL USE (lb)	EI	POUNDS NOX
7000 <sup>a</sup>	640	1050	40	700.0	2.26	.1.58
11524	8697	5383	6	538.3	9.33	5.03
11744	10226	6354	6	635.4	12.22	7.76
12022	11764	7474	12	1494.8	16.02	23.94
12333	13393	8731	7	1018.6	21.33	21.73
11359	8688	5457	10	909.5	9.32	8.48
11726	10167	6311	10	1051.8	12.09	12.72
12063	11933	7634	10	1272.3	16.50	20.99
12800 b	12800	8040	15	2010.0	19.22	38.63
						<del></del>
Pounds	of fuel u	used in test		9630.8		
Pounds of NOx per test						

<sup>(</sup>a) An rpm of 7000 indicates that the engine is operating at idle. The corresponding thrust and fuel flow are estimated.

<sup>(</sup>b) An rpm of 12800 indicates that the engine is operating at military. The corresponding thrust and fuel flow are estimated.

TABLE 15. Emission of oxides of nitrogen from the testing of a TF41 engine at NAS Lemoore (Engine Serial Number: 141357)

Date:	9/16/198	35						
RPM	THRUST (1b)	FUEL FLOW (lb/hr)	TIME (MIN)	FUEL USE (lb)	EI	POUNDS NOX		
7000	640	1050	60	1050.0	2.26	2.37		
11607	8940	5579	5	464.9	9.74	4.53		
11767	10275	6453	5	537.7	12.32	6.63		
12079	11943	7613	10	1268.8	16.53	20.97		
12391	13557	8809	6	880.9	21.96	19.34		
11582	8857	5551	5	462.6	9.60	4.44		
11843	10530	6644	5	553.7	12.89	7.14		
12061	11963	7653	5	637.7	16.59	10.58		
12492	19142	9381	6	938.1	58.68	55.05		
12800 b	12800	8040	20	2680.0	19.22	51.51		
						<del></del>		
Pounds of fuel used in test 9474.5								
Pounds	Pounds of NOx per test 182.55							

<sup>(</sup>a) An rpm of 7000 indicates that the engine is operating at idle. The corresponding thrust and fuel flow are estimated.

<sup>(</sup>b) An rpm of 12800 indicates that the engine is operating at military. The corresponding thrust and fuel flow are estimated.

TABLE 16. Emission of oxides of nitrogen from the testing of a TF41 engine at NAS Lemoore (Engine Serial Number: 141551)

Date: 8/20/1985

RPM	THRUST (1b)	FUEL FLOW (lb/hr)	TIME (MIN)	FUEL USE (lb)	EI	POUNDS NOX
7000 a	640	1050	70	1225.0	2.26	2.77
11564	8744	5385	4	359.0	9.41	3.38
11801	10250	6391	4	426.1	12.27	5.23
12078	11803	7479	9	1121.8	16.13	18.09
12497	13822	9139	6	913.9	23.01	21.03
11545	8691	5358	5	446.5	9.33	4.16
11796	10329	6387	5	532.2	12.44	6.62
12088	11831	7540	5	628.3	16.21	10.18
12603	14587	9647	7	1125.5	26.32	29.63
12800 b	12800	8040	25	3350.0	19.22	64.38
				<del></del>		<del></del>
Pounds	of fuel	used in test	10128.4			

Pounds of NOx per test

165.47

Pounds of NOx per pound of fuel used in test = 0.01634

<sup>(</sup>a) An rpm of 7000 indicates that the engine is operating at idle. The corresponding thrust and fuel flow are estimated.

<sup>(</sup>b) An rpm of 12800 indicates that the engine is operating at military. The corresponding thrust and fuel flow are estimated.

TABLE 17. Emission of oxides of nitrogen from the testing of a TF41 engine at NAS Lemoore (Engine Serial Number: 141511)

Date:	8/ 5/198	35				
RPM	THRUST (lb)	FUEL FLOW (lb/hr)	TIME (MIN)	FUEL USE (lb)	EI	POUNDS NOX
7000 a	640	1050	100	1750.0	2.26	3,96
11632	8862	5467	7	637.8	9.61	6.13
11845	10210	6336	7	739.2	12.18	9.01
12005	11454	7137	13	1546.3	15.17	23,45
12193	12705	8045	7	938.6	18.90	17.74
11512	8140	507.7	6	507.7	8.46	4.30
11695	9327	5772	6	577.2	10.43	6.02
11889	10575	6650	6	665.0	12.99	8.64
12042	11819	7506	6	750.6	16.17	12.14
12800 b	12800	8040	30	4020.0	19.22	77.26
Pounds of fuel used in test				12132.5		
Pounds of NOx per test				-		168.64

<sup>(</sup>a) An rpm of 7000 indicates that the engine is operating at idle. The corresponding thrust and fuel flow are estimated.

<sup>(</sup>b) An rpm of 12800 indicates that the engine is operating at military. The corresponding thrust and fuel flow are estimated.

TABLE 18. Emission of oxides of nitrogen from the testing of a TF41 engine at NAS Lempore (Engine Serial Number: 141612)

Date:	8/ 6/1985						
RPM	THRUST (lb)	FUEL FLOW (lb/hr)	TIME (MIN)	FUEL USE (lb)	EI	POUNDS NOX	
7000 a	640	1050	55	962.5	2.26	2.18	
11518	8793	5405	7	630.6	9.49	5.99	
11811	10406	6541	7	763.1	12.61	9.62	
12096	11998	7573	14	1767.0	16.69	29.49	
12453	13677	8955	7	1044.7	22.43	23.43	
11537	8796	5439	6	543.9	9.50	5.17	
11836	10603	6640	7	774.7	13.06	10.11	
12108	12075	7641	7	891.5	16.92	15.08	
12481	13805	9058	7	1056.8	22.94	24.24	
ь 12800	12800	8040	30	4020.0	19.22	77.26	
Pounds of fuel used in test 12454.8							
Pounds of NOx per test 202.							

<sup>(</sup>a) An rpm of 7000 indicates that the engine is operating at idle. The corresponding thrust and fuel flow are estimated.

<sup>(</sup>b) An rpm of 12800 indicates that the engine is operating at military. The corresponding thrust and fuel flow are estimated.

TABLE 19. Emission of oxides of nitrogen from the testing of a TF41 engine at NAS Lemoore (Engine Serial Number: 141630)

Date:	8/24/198	35				
RPM	THRUST (lb)	FUEL FLOW (lb/hr)	TIME (MIN)	FUEL USE (lb)	EI	POUNDS NOX
7000 <sup>a</sup>	640	1050	30	525.0	2.26	1.19
11547	8710	5376	5	448.0	9.36	4.19
11782	10276	6430	5	535.8	12.33	6.60
12054	11839	7535	9	1130.2	16.23	18.34
12502	14027	9364	5	780.3	23.85	18.61
11546	8765	5405	5	450.4	9.45	4.26
11779	10280	6451	5	537.6	12.33	6.63
12149	12206	7927	5	660.6	17.31	11.44
12800 b	12800	8040	20	2680.0	19.22	51.51
				<del></del>		
Pounds of fuel used in test 7748.0						
Pounds of NOx per test						122.77

<sup>(</sup>a) An rpm of 7000 indicates that the engine is operating at idle. The corresponding thrust and fuel flow are estimated.

<sup>(</sup>b) An rpm of 12800 indicates that the engine is operating at military. The corresponding thrust and fuel flow are estimated.

TABLE 20. Emission of oxides of nitrogen from the testing of a TF41 engine at NAS Lemcore (Engine Serial Number: 141952)

Date:	8/ 2/198	35				
RPM	THRUST (lb)	FUEL FLOW (lb/hr)	TIME (MIN)	FUEL USE (lb)	EI	POUNDS NOX
7000 <sup>a</sup>	640	1050	60	1050.0	2.26	2.37
11420	8687	5407	5	450.6	9.32	4.20
11584	9808	6112	5	509.3	11.35	5.78
11743	10722	6728	8	897.1	13.33	11.96
11966	11802	7573	5	631.1	16.12	10.17
12800 <sup>b</sup>	12800	8040	10	1340.0	19.22	25.75
						<del></del>
Pounds	of fuel u	used in test		4878.1		
Pounds	of NOx pe	er test				60.24

Pounds of NOx per pound of fuel used in test = 0.01235

<sup>(</sup>a) An rpm of 7000 indicates that the engine is operating at idle. The corresponding thrust and fuel flow are estimated.

<sup>(</sup>b) An rpm of 12800 indicates that the engine is operating at military. The corresponding thrust and fuel flow are estimated.

TABLE 21. Emission of oxides of nitrogen from the testing of a TF41 engine at NAS Lemcore (Engine Serial Number: 141952)

Date:	8/13/1985					
RPM	THRUST (lb)	FUEL FLOW (lb/hr)	TIME (MIN)	FUEL USE (lb)	EI	POUNDS NOX
7000 <sup>a</sup>	640	1050	50	875.0	2.26	1.98
11414	8562	5366	6	536.6	9.12	4.89
11678	9855	6251	7	729.3	11.45	8.35
11925	10974	7060	14	1647.3	13.94	22.96
12267	12717	8388	7	978.6	18.94	18.53
12800 <sup>b</sup>	12800	8040 .	20	2680.0	19.22	51.51
				<del></del>		
Pounds	of fuel u	ısed in test		7446.8		
Pounds	of NOx pe	er test				108.22

Pounds of NOx per pound of fuel used in test = 0.01453

<sup>(</sup>a) An rpm of 7000 indicates that the engine is operating at idle. The corresponding thrust and fuel flow are estimated.

<sup>(</sup>b) An rpm of 12800 indicates that the engine is operating at military. The corresponding thrust and fuel flow are estimated.

TABLE 22. Emission of oxides of nitrogen from the testing of a TF41 engine at NAS Lemoore (Engine Serial Number: 141952)

Date: 9/23/1985

RPM	THRUST (lb)	FUEL FLOW (lb/hr)	TIME (MIN)	FUEL USE (lb)	EI	POUNDS NOX
7000 <sup>a</sup>	640	1050	75	1312.5	2.26	2.97
11403	8458	5257	5	438.1	8.95	3.92
11646	9764	6150	5	512.5	11.26	5.77
11890	10942	6941	10	1156.8	13.86	16.03
12242	12689	8230	5	685.8	18.85	12.93
11561	9338	5842 .	9	876.3	10.45	9.16
11766	10266	6534	9	980.1	12.30	12.06
11979	11382	7242	9	1086.3	14.97	16.27
12800 <sup>b</sup>	12800	8040	35	4690.0	19.22	90.14
Pounds	of fuel	used in test		11738.5		

Pounds of NOx per test

169.24

Pounds of NOx per pound of fuel used in test = 0.01442

<sup>(</sup>a) An rpm of 7000 indicates that the engine is operating at idle. The corresponding thrust and fuel flow are estimated.

<sup>(</sup>b) An rpm of 12800 indicates that the engine is operating at military. The corresponding thrust and fuel flow are estimated.

TABLE 23. Emission of oxides of nitrogen from the testing of a TF41 engine at NAS Lemoore (Engine Serial Number: 141952)

Date:	9/27/198	35				
RPM	THRUST (1b)	FUEL FLOW (lb/hr)	TIME (MIN)	FUEL USE (lb)	EI	POUNDS NOX
7000 <sup>a</sup>	640	1050	120	2100.0	2.26	4.75
11480	8739	5480	5	456.7	9.40	4.29
11787	10353	6447	5	537.2	12.49	6.71
12105	11847	7468	9	1120.2	16.25	18.20
12578	14175	9353	7	1091.2	24.48	26.71
11483	8775	5424	5	452.0	9.46	4.28
11776	10269	6365	5	530.4	12.31	6.53
12075	11752	7417	12	1483.4	15.98	23.71
12404	13181	8569	7	999.7	20.55	20.55
11466	8689	5417	6	541.7	9.32	5.05
11767	10264	6371	6	637.1	12.30	7.84
12090	11810	7461	6	746.1	16.15	12.05
12686	14693	9725	7	1134.6	26.82	30.43
12800 b	12800	8040	20	2680.0	19.22	51.51
Pounds	of fuel	used in test		14510.3		<del></del>
Pounds	of NOx p	er test				222.60

Pounds of NOx per pound of fuel used in test = 0.01534

<sup>(</sup>a) An rpm of 7000 indicates that the engine is operating at idle. The corresponding thrust and fuel flow are estimated.

<sup>(</sup>b) An rpm of 12800 indicates that the engine is operating at military. The corresponding thrust and fuel flow are estimated.

TABLE 24. Emission of oxides of nitrogen from the testing of a TF41 engine at NAS Lempore (Engine Serial Number: 142563)

Date:	8/15/198	35				
RPM	THRUST (lb)	FUEL FLOW (lb/hr)	TIME (MIN)	FUEL USE (lb)	EI	POUNDS NOX
11428	8802	5411	5	450.9	9.51	4.29
11804	10360	6568	5	547.3	12.51	6.85
12037	11922	7548	10	1258.0	16.47	20.72
12518	14378	9553	. 6	955.3	25.37	24.24
11512	8753	5365	5	447.1	9.43	4.21
11790	10351	6523	7	761.0	12.49	9.50
12067	11946	7622	7	839.2	16.54	14.71
12555	14536	9625	6	962.5	26.09	25.11
11527	8743	5435	5	452.9	9.41	4.26
11782	10457	6531	6	653.1	12.72	8.31
12046	11876	7597	12	1519.4	16.33	24.82
12342	13393	8764	5	730.3	21.33	15.58
11553	9023	5521	6	552.1	9.89	5.46
11775	10384	6527	6	652.7	12.56	8.20
12046	11930	7614	6	761.4	16.49	12.56
12492	14143	9470	6	947.0	24.34	23.05
Pounds	of fuel	used in test		12540.3		
Pounds	of NOx p	er test				211.86

Pounds of NOx per test

211.86

Pourds of NOx per pound of fuel used in test = 0.01689

<sup>(</sup>a) An rpm of 7000 indicates that the engine is operating at idle. The corresponding thrust and fuel flow are estimated.

<sup>(</sup>b) An rpm of 12800 indicates that the engine is operating at military. The corresponding thrust and fuel flow are estimated.

TABLE 25. Emission of oxides of nitrogen from the testing of a TF41 engine at NAS Lemoore (Engine Serial Number: 142563)

Date:	8/15/198	35				
RFM	THRUST (lb)	FUEL FICW (lb/hr)	TIME (MIN)	FUEL USE (lb)	EI	POUNDS NOX
7000 <sup>a</sup>	640	1050	180	3150.0	2.26	7.12
11501	8762	5379	6	537.9	9.44	5.08
11812	10558	6653	6	665.3	12.95	8.62
12056	11908	7605	11	1394.2	16.43	22.90
12525	14315	9538	7	1112.8	25.09	27.92
11520	8781	5403	6	540.3	9.47	5.12
11767	10397	6483	6	648.3	12.59	8.16
12071	11927	7673	6	767.3	16.48	12.65
12453	13975	9253	8	1233.7	23.63	29.16
11479	8745	5343	5	445.2	9.41	4.19
11762	10406	6480	5	540.0	12.61	6.81
12060	11992	7642	5	636.8	16.67	10.62
12584	14565	9775	6	977.5	26.22	25.63
11530	8899	5483	5	456.9	9.67	4.42
11762	10418	6516	5	543.0	12.64	6.86

Emission index (EI) for NOx is expressed as pounds of NO  $_{\rm 2}$  per 1000 pounds of fuel.

12078 11980 7682 6 768.2 16.64 12.78

<sup>(</sup>a) An rpm of 7000 indicates that the engine is operating at idle. The corresponding thrust and fuel flow are estimated.

<sup>(</sup>b) An rpm of 12800 indicates that the engine is operating at military. The corresponding thrust and fuel flow are estimated.

TABLE 25 (continued)

RPM	THRUST (lb)	FUEL FLOW (lb/hr)	TIME (MIN)	FUEL USE (lb)	EI	POUNDS NOX
12581	14525	9823	7	1146.0	26.04	29.84
11543	8935	5492	6	549.2	9.73	5.35
11780	10398	6546	6	654.6	12.59	8.24
12062	11970	7681	6	768.1	16.61	12.76
12438	13864	9230	7	1076.8	23.18	24.96
12800 b	12800	8040	20	2680.0	19.22	51.51
Pounds	of fuel	used in test		21292.3		
Pounds	of NOx p	er test				330.69

Pounds of NOx per pound of fuel used in test = 0.01553

<sup>(</sup>a) An rpm of 7000 indicates that the engine is operating at idle. The corresponding thrust and fuel flow are estimated.

<sup>(</sup>b) An rpm of 12800 indicates that the engine is operating at military. The corresponding thrust and fuel flow are estimated.

TABLE 26. Emission of oxides of nitrogen from the testing of a TF41 engine at NAS Lemoore (Engine Serial Number: 142596)

Date: 10/21/1985

RPM	THRUST (lb)	FUEL FLOW (lb/hr)	TIME (MIN)	FUEL USE (lb)	EI	POUNDS NOX
7000 <sup>a</sup>	640	1050	75	1312.5	2.26	2.97
11483	8600	5360	5	446.7	9.18	4.10
11727	9891	6231	5	519.2	11.52	5.98
12056	11651	7401	10	1233.5	15.73	19.40
12545	14136	9402	5	783.5	24.31	19.05
11440	8500	5262	5	438.5	9.02	3.95
11706	9817	6171	5	514.2	11.37	5.85
12019	11440	7296	5	608.0	15.13	9.20
12741	15022	10177	8	1356.9	28.42	38.56
ъ 12800	12800	8040	25	3350.0	19.22	64.38
				<del> </del>		
Pounds	of fuel	used in test		10563.1		

Pounds of NOx per test

173.44

Pounds of NOx per pound of fuel used in test = 0.01642

<sup>(</sup>a) An rpm of 7000 indicates that the engine is operating at idle. The corresponding thrust and fuel flow are estimated.

<sup>(</sup>b) An rpm of 12800 indicates that the engine is operating at military. The corresponding thrust and fuel flow are estimated.

TABLE 27. Emission of oxides of nitrogen from the testing of a TF41 engine at NAS Lemoore (Engine Serial Number: 142599)

Date: 11/15/1985

RPM	THRUST (1b)	FUEL Flow (lb/hr)	TIME (MIN)	FUEL USE (lb)	EI	POUNDS NOX
7000 <sup>a</sup>	640	1050	50	875.0	2.26	1.98
11556	8672	5557	7	648.3	9.29	6.03
11759	10182	6516	8	868.8	12.12	10.53
12077	11757	7711	12	1542.2	16.00	24.67
12485	13756	9314	7	1086.6	22.74	24.71
11556	8650	5403.	5	450.2	9.26	4.17
11817	10279	6494	7	757.6	12.33	9.34
12090	11790	7610	7	887.8	16.09	14.28
12587	14218	9585	10	1597.5	24.67	39.41
12800 <sup>b</sup>	12800	8040	30	4020.0	19.22	77.26
Pounds	of fuel	12734.2				
Pounds	of NOx p	er test				212.38

Pounds of NOx per pound of fuel used in test = 0.01668

<sup>(</sup>a) An rpm of 7000 indicates that the engine is operating at idle. The corresponding thrust and fuel flow are estimated.

<sup>(</sup>b) An rpm of 12800 indicates that the engine is operating at military. The corresponding thrust and fuel flow are estimated.

TABLE S2. Statistical summary of emissions data for the TF41-A-2B engine

Table	Pounds of NOx per test	Pounds of NOx per test (calculated)	Difference d (a)	d <sup>2</sup>
1	72.06	77.26	-5.20	27.04
2	54.38	61.92	<del>-</del> 7.54	56.85
3	214.77	220.87	-6.10	37.21
4	111.37	106.63	4.74	22.47
5	42.03	41.56	0.47	0.22
6	122.26	122.67	-0.41	0.17
7	111.45	. 117.25	<del>-</del> 5.80	33.64
8	114.96	121.49	<b>-6.</b> 53	42.64
9	185.04	198.18	-13.14	172.66
10	192.06	218.58	-26.52	703.31
11	524.00	509.57	14.43	208.22
12	473.31	464.74	8.57	73.44
13	338.98	339.87	-0.89	0.79
14	140.86	145.91	-5.05	25.50
15	182.55	143.54	39.01	1521.78
16	165.47	153.45	12.02	144.48
17	168.64	183.81	-15.17	230.13
18	202.57	188.69	13.88	192.65
19	122.77	117.38	5.39	29.05
20	60.24	73.90	-13.66	186.60
21	108.22	112.82	-4.60	21.16
22	169.24	177.84	-8.60	73.96
23	222.60	219.83	2.77	7.67
24	211.86	189.99	21.87	478.30
25	330.69	322.58	8.11	65.77
26	173.44	160.03	13.41	179.83
27	212.38	192.92	19.46	378.69
totals	5028.20	4983.28	44.92	4914.25
means	186.23	184.57	1.66	

Estimated standard deviation = 13.64

Estimated standard error = 2.63

95% confidence limits = ±5.40

Relative accuracy = 3.79%

(a) Difference = NOx (per test) - NOx (calc'd)

See notes i - iv.

Notes:

i. Estimated standard deviation:

$$S_{d} = \sqrt{\frac{\Sigma d^{2} - (\Sigma d)^{2}/n}{n-1}}$$

ii. Estimated standard error:

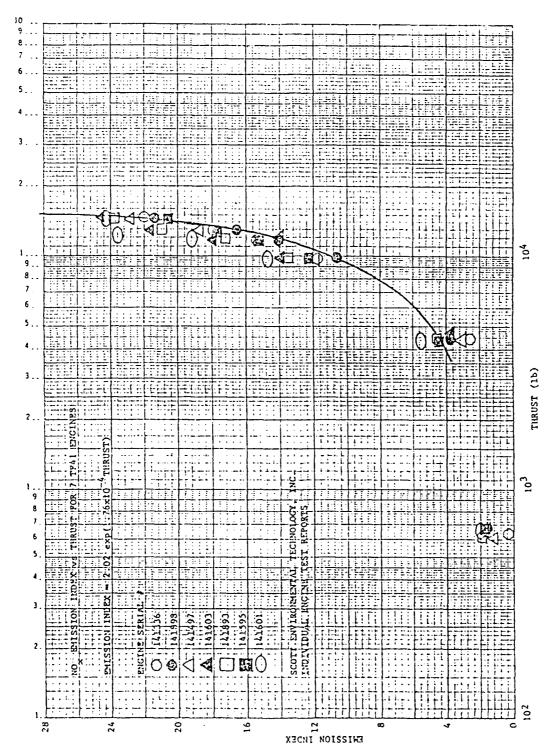
$$S_{\overline{d}} = \frac{S_{\overline{d}}}{\sqrt{n}}$$

iii. 95% confidence limits (two-tailed test):

$$CC = t_{0.95} S_{d}$$
 where  $t_{0.95} = 2.056$  for  $n = 27$ 

iv. Relative accuracy:

$$RA = \frac{|\overline{d}| + |CC|}{RM} \times 100$$
 where  $\overline{RM}$  is the average "reference method" value



igure 1. NO EMISSION INDEX vs THRUST FOR 7 TF41-A-2B ENGINES

APPENDIX A: Engine Operational Data (Run Sheets)

## ## TF41-A-2B "NAS Lemove, CA

	1			
THRUST	WH RAM	FUEL Flow	Total Time	Eng SER#
8 :10	19001	5578	6 Mis	141011
adgo	12187	6346	6 1	7-7-87
1,380	12427	746a	5	
12440	12645	8343	6	
8 ao	11958	5351	5	
1000	12830	6483	5	
11-160	13480	754a	5	
15 480	12694	8345	6 Min	
3550	11991	5525	5 Min	141505
₹ 50	12192	6410	5 1	7-9-87
11480	12531	7614	5	
17 520	12786	8557	5	
8550	13012	5625	6	
יביילס'	12225	6525	5	
11 70	12543	7665	5 M.s	
8740_	11945	5630	5 Min	141619
16 60	lalleo	6500	6 1	7-10-87
11460	12467	7.561	6	
2 19O	12618	8165	6	
8660	11934	5491	6	
00 0	13108	6520	S	
11/430	12469	7589	7	
2470	12712	8486	6	
8. 20	11945	5465	5	
10180	1aa39	6663	5	
11 90	12506	7453	5 V	
12590	12733	8526	6 M.N	
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7.	1	1		
IMRUST	NHRPM	FUEL Flow	Total Time	ENG SER#
8570.	11712.	5281	5 Miss.	141430 6-23-87
10010	११५७५	6243	5 Mis	,
1540	12334	7280	5 Min	
13450	12815	orop	5 M.N	
3430	11739	5246	5 Min	
12060	12030	le3le3	5 Min	
1430	13308	7375	5 Min	-
.3310	19684	8793	5 Min	
¥5∞	11895	5381	6 Min	
770	12144	6363	7 Min	
11380	12435	7415	7 Min	
3960	12780	8758	5 Min	
9500	11954	5409	8 Mid	
12030 .	13308	4040	7 Min	
1.350	12472	7418	8 Miss	
12880	12779	8633	7_Min	
080	12016	5545	6 Min	
10160	13948	6553	8 M:w	
.470	12522	7550	7 Min	
12730	12788	8586	9 M:1	
2.30	11743	5959	8 Mis	141011 5.21.87
10130	11979	6639	8 Miss	
n500	)13918	7723	7 Min	
1:270	12596	8847	9 Min	
8550	11738	5356	7 Min	
1,170	ISOSS	6449	6 Min	
11580	12274	7457	6 Min	
1,090	12811	9542	7 Wis	
4				,
1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1				
-	1	1	<u> </u>	

			, 1	
Thrust	NH RPM	TOEL FLOW	TOTAL TIME	ENG SERT
,430	11804	5320	5 mid	143542 6.14.8
10380	12124	<u>6586</u>	6 mid.	
.1500	12364	7483	5 mil	
13030	12686	8724	7 mis.	
3760	11707	5479	5 Min	141483 6.18.87
10160	11993	6398	5 1	
11410	19 328	7313	6	
3210	12651	8823	5	
8470	11707	5310	6	
303o	12028	(6381	7	
113-10	12319	7357	6	
\$3 <del>7</del> 0	12742	8962		
F430	11772	553 <u>0</u>	6	
_OFOC	12055	6371	5	
11510	19344	7441	6	_
13300	12767	9038	6 Min	
. '270	11905	5651	5 Min	141931
10010	12097	6436	5 1	6-24-87
нō	12348	7389	4	
13050	12697	8827		
্যুহত	11945	5555	5	
10090	12185	6608	5	
-15 <u>2</u> 0	19ने १।	7537	6	_
3300	12420	8200	6	
8570	11925	5459	7	
1070	12177	6551	. 5	
11550	12438	7668	5	
1.690	12644	8493	6 Min	
	`,		•	
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THRUST	NA RPM	Fuel Flaw	Total Time	Eng Sex#
8450	11849	5271	7 Min	141359
10010	18183	4390	7 1	7-15-87
11350	18490	7410	6	
. la350	12708	8930	8	
8500	每11930	5400	6	
9990	19162	6375	7.	
11470	12482	7485	7	
lgieo	12646	8020	7	
8550	11975	5440	7	
10000	12,430	6503	7	
11450	12525	7545	8	
12550	12760	8435	7	
8540	11985	5430	7	
9950	12250	6420	6	
11450	12545	7515	8 /	
12610	12790	8465	8 Min	
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TF-41 PERFORMANCE CALCULA	TION REC	ORD			1439	177	x /2/2 E TEST	
ENGINE SERIAL NUMBER /4/	1787	GEOLE	SVOC VARA		132		A PLATE	
			ENCE NUMB		2.6.6.1.1		1+16	<del>///</del>
TEST CELL MARER TW		OPERA		4777		I IS PECT	( E S S )	3/120
	Bingo	TO BE	STCRED/RI	Panyl	600 S	407/2	$\sim$	7.00
DATA	1	2	3	4	<u> </u>		<del></del>	<del></del>
DEM BOINT	56	56	56	56	59	59	54	59
PANS	1				T		13.60	·
P)	27-61	29.58		29.53.	29.47	29.41	22.39	29.33
Tl	34.	84.	٠,۲۶	85.	/00.	101.	10%.	102.
TF (Fuel Temp)	88	93.	76.	98.	8/.	13.	96.	8.
. AL AREA			43	. 798			<del></del>	
LAB SG	• 32	570 6	0°	<del></del>	. દુ: ગુ	)4 3 (	60°	·
P3	362	795	442	502	372	425	483	504
Т3	744	725	518	873	786	838	859	911
RES. VALUE					4.360	1.360	4.360	4.360
J BCX TEMP (JBT)			$\overline{}$		144.	148.	14.2	131.
FIIO Obs	8170	9150	10520	12720	8400	9970		12730
FIIC Thrust	8304	9320	10741	12612	8572	10,201	1/865	13545
FNT	8415	l .	10351	1		10311	11976	12656
77°	4436	į.	10379	l		10537		12687
NILO Obs	7299	7571	7934	8410	7489	7920		5547
NIC	7114	7379	7726	8190	7192	7600	8018	899
1HO 0bs	11611	11790	12077			12134	<del></del>	17622
1710 (C.C.22)	11318	11192	11763	12116	1/355	11643	11972	12/0/
. WFO Obs	5088	5745	67/1	8052	5347	6450	7670	8160
WTC Lieb Maw.	5152	5814	6791	8/61	5320	621	7634	8/30
WFC 77°				~				
P5.1 Obs	77.9	25.8	30.2	35.7	23.9	19.0	34.0	36.0
P5C	53.60	56.67	61.31	67.11	54.75	6013	65.40	67.53
E.P.R.	1.791	1.874	2.049	2.243	1.830	2009	2561	2.257
DELTA P Cos	38.8	43.6	50.2	59.2	40.3	479	56.5	54.3
WAIC	204.2	214.1	227.1	243.4	207.9	273 .3	239.4	2443
T5 Obs	984	1032	1093	1165	916	982	1042	1069
T5C 77°	967	1015	1072	1143	864	925	981	1004
тыQSC	1739	1812	1909	2043	1749	1867	1977	2027
TLQS (A)	1767	1841	1944	2086	1844	1971	2010	2143
START TIME:	<del></del>	TIME:		Τσ	TAL RUN	<u> የ የላይ</u> :	hrs.	min.

F-41 PERFORMANCE CALCULA	TION REC	סאכ		TSH L	14314 1379 432	TYPE	1227 1881 P.	
FIGURE SERIAL NUMBER /4//	2 6 7	SEQUE	ENCE NUMS			DATE /	<u>`````</u>	~~~ ~~
	0		ntar Sq.					
1101 00110 110 110 110			STCRED/RE			110.07.0	.00-4	38
DATA			124/2		5	6	7	8
DFW POINT	ļ		5.4					-
PANS		,	79.72	1				:
P1			29.50			•	•	-
					<u> </u>			<b>-</b> -
TF (Fuel Temp)	99	0,	78. 94.	G/	•	•	<u> </u>	
	71.	7/•			<u> </u>	<u> </u>	•	<u> </u>
AL AREA	(10	5 0 60		798	J		0	
IAB SO			1	r . v	•	<u> </u>	<u> </u>	γ
P3	·		47.5					 
т3	756	8/80	345	8 19	11.000			
RES. VALUE	$\rightarrow$	<del></del>			41252	4.252	4,752	4.75
J BCX TEMP (JBT)	2660		•	-	•	•	•	•
	1	1	11540	1				\
FIC. Thuset		Ţ		1				<u> </u>
FNT			11937	13202		<del> </del>		
FNT 77 <sup>0</sup>	8895	1		13235				
MLO Obs		-	8170	-				
nc	<del></del>		8009					
1510 Obs	11650	12.035	12250	12495				
MIC FAREST IPM	11450	11798	12009	15225				
, WFO Cbs	5390	6643	7396	8375				
WFC fuel flow.	547,52	6762	7531	8553				
WFC 77°								-
P5.1 Obs	74.4	30.3	33.4	374	•			
PSC	55.25	61.43	154.73	68.95	•		•	•
E.P.R.	1.545	2.053	2.163	7.305	•			
DELTA P Cos	11.4	50.1	55.4	61.2	•	•		•
WAIC			2%.9					
T5 0bs		1	1128					
т5с 77°	1004	1086	1125	1179				
TliQSC		1919	1990	·		1	1	
Tligs (A)			1495					
START TIME:		TIME:	<del></del>		rat, Run	ፓ [ ME :	hrs.	min.

INTE TIME 2 LOURS 30 MINTO Additional accounty Page - Domine.

TF-41 PERFORMANCE CALCULA	TION REC	ORD		FNC CK TSN TSO =	3635			TEST PLATE	
ENGINE SERIAL NUMBER /41	790	SEQUE	ENCE NUMB		89				
	<u>C.10</u> Z.							2 ADI H	
			STCRED/RE		12-040			· ADI H	1015616
DATA	1	2	3	lı	5	6	<del></del>	7	В
DEW POINT	60	7000	14111N		901	311,	W	400,00	61
РАМВ								29.64	
PJ	1				T			29.42	
71		I	33. 7					84.9	
TF (Fuel Temp)	}	93.			93.	95.			<u> </u>
Al; AREA			355		<u> </u>	·		1-2-	-!
IAB SG	. 87	190 E		<u> </u>	٠:	2 SG		,0°	
P3	277	425	1	507	333			473	533
T3	<del>                                     </del>	806		384	772	<del> </del>		854	909
RES. VALUE								4.629	
J BCX TEMP (JBT)					107	110			115.
FMO Obs	3520	10150	11110	12450				11500	
INO Thurst 1								11800	
FINT	1	1	11553			1		1	
FRT 77°	I	i		i				11950	
NLC Cbs	7493	1.	1		1	1		3275	
NLC	7307		7955					8055	
iiiO Opa	·	1			1	1		12340	
IHC (1.12/22) /							-	12:012.	
WFO Cos	[	LAZ3	1	800	1				
WFC hul Maw	5531	6519	7299			7		7541	€'`
W7C 77°						-		-	
P5.1 Obs		28.8				7.3.	سې,	33.1	38. 7
P5C			63.29		_	1			-
E.P.R.						T		2.156	1
DELTA P Cos		}	1		1	i —		55.0	
WAIC	;		233.8						
T5 Obs		1068		1168		1		970	107
т5с 77°	994	1054		1149	363	Ī		953	1006
TliQSC		1314	1965	l	[	i		1910	
TliQS (A)	1821	1919	1974		1823				
START TIME:	STOP	TIME:			TAL PUN	*****	بببالنس	hrs.	min.

Additional military 12th This 35 mil I'Dle Time Ihr 10min

F-41 PERFORMANCE CALCU	LATION REC	ORD			635		TYPE	TEST F-	/ <u>C</u>
ENGINE SERIAL NUMBER	41248	SEOU	ENCE MIM	BER 78	730 a	DATE			
TEST CELL NUMBER 2			ATORAD 1			i		ALIG	
7. 100 III III X			STORED/RI	· · · · · · · · · · · · · · · · · · ·		1110.1	3.71 CI	R NDa	F.CICI'
DATA	6:40		6.11/2		5	T	<del></del>	7	8
DEW POINT	60	60	<del> </del>	60		<del>                                     </del>			-
PAMB	<del>-  </del>	1	29.66		•	<del>                                     </del>			<b>-</b>
P1		1	1	29.42.		<del>                                     </del>			
T1		1	76.				,		
TF (Fuel Temp)	95.	i	i .	98.	•		·		1.
Ali AREA		, 855						<u></u>	
IAB SG	0.83	47 @ 6	0 °		•	6		0	
Р3	374	447	473	565	-				
т3		813	836	917			·		1
RES. VALUE	6.009	6.009	6.009						1
J BCX TEMP (JBT)	102 •	104 •	104 •	106 •	•	T.	•		
FNO Obs	7500	10675	11525	1.3980					
FIC Thunt-	8691	10944	11841	14412					
FNT	7801	11059	11951	14522					
FNT 77°	8420	11087	11981	14559					
ILO Obs	7405	8000	8190	2870					
MC	7262	7845	7039	8707					
MiO Cbs	111,30	12082	12245	12755					
инс дет	11405	11849	12020	12520		_			
WFO Obs	5319	6855	7430	94.25	******		- Transfer	,	
WEC Level flow	5426	7010	7616	9688				<u> </u>	
WFC 77° / -	_	<u> </u>							
P5.1 Obs	29 • 1	30.8	33 • 3	41.3		-	•		<del>  .</del>
PSC	55 • 00	<u> </u>	64.72	73 . 19	•	-	•	<u>  •  </u>	<u>  •</u>
E.P.R.	1.838	1.073	2.163	2.446	•	<u>  .</u>	<del></del> -	ļ	·
DELTA P Obs	11.0	51.6	55 • 6	67.6	•		•	·	<u> </u>
WAIC	2.69 • 3		237.5	257.8	•	<del> </del>	•	·	<b>-</b>
T5 Obs	9,72	952	977	1065		<del> </del>			-
T50 77°	874	154	981	1069		-		ļ	-
ThQSC	1775	1944	2006	2143					-
TliQS (A)	1775	1944 TIME:	3001	2178		<u> </u>			

. .

INE TIME Thour 5 min Additional Militing Time is with

TF-41 PERFORMANCE CALCULA	TION REC	ORD		FIC CX TSH TSO =	3635			TEST PLATE	
ENGINE SERIAL NUMBER	790	SEOUE	ENCE NUME	4	2730	DAME:			
	<u>c. 13                                    </u>			<u>-</u>	01	T HS DE	<u>ו</u>	<u>- Aug :</u>	<u> </u>
			STORED/RE		12700 [			3 ADI H	- AND STEE
DATA	1	2	3	h	5	6	······	7	В
DEW POINT	3711.0	7.21.0	140110		61	777.	·//	7.11, N	17170
FANG		29.17				1		29.64	75 14
F)	1							29.42	
Tl	1					j		84.9	
TF (Fuel Temp)	1		93.			25.			15.
. Al. AREA			355		L			<del></del>	
IAB SG	.87	19 e 6			• 😅	/		, 0°	<del></del>
P3	1	ì	453	507	383	-			553
т3			840					854	909
RES. VALUE	X			><	I	1		4,629	
J BOX TEMP (JBT)	X		X	><	!	1113			115•
FNO Cus	3530	10150	111160	1245C	î	1		11500	
FINO Think				1		1		11839	
FRT	1	l	İ			1		11920	
FINT 77°	i	1	1	1	)			11950	
NLO Obs	1	1	i			i		3275	
NLC						1		8055	
M10 059	11740								
IHC 27 (2.27)						1		12012	
. WTO Obs	1	6428	i	800				1	8775
WFC hul Maw	5531	6519	7299	3-107-	5599	34	రేస	7541	<b>ਦੇ</b> ਵਰਤ
WFC 77°							5	5	
P5.1 Obs	24.5	28 • 8	32.0	51. 2		ر ج	بنغ.	33.1	38.3
PSC	55.37	59.95	63.29	6.7.72.	55 <b>.</b> g-	54.	97	64.51	70.05
E.P.R.	1.850	Z• 003	2.115	2.2.63	1.867	ن . ?	:01	2.156	2.3-90
DELTA P Cos	1		53.60	İ	1	1	-		63.0
WAIC	210.6	222.6	233+8	245.6	212.4	2.7	رب ورد		<u>250. 4</u>
T5 Chs		1063		1168		1		970	1025
TSC 77°	994	1054	1014	1149		ر.۱:۰		953	IONE
ThQSC	1796	1314	1965		1793				2112
Thos (a)	1831	1919	1994		1823	1		2026	
START TIME:	STOP	TIME:		то	TAL RUN	य । धन्त्रः स	:	hrs.	min.

IPIE TIME THE 10 MIN AMITICIAL MILITARY PORT THE 35 MINE

TF-41 PERFORMANCE CALCULA	ATION RECO	PRD		TSI TSO	1439	TYPE		
ENGINE SERIAL NUMBER /4	(1234	SEQUE	NCE NUMB	BER 8	00	DATE 6	OCT 8	?S
TEST CELL NUMBER		OPER/	TOR Z	RISAG	$\overline{}$	INS PECTO		
	DATA		TCRED/RE		<del></del>	<del></del>	<u></u>	
DATA	Sindrato	g mishires	3	d winers	18 01/24 (10) 5	3 mi vares	711.000	711. Ucle
DEW POINT	51	51	51	51	51	5/	51	31
PAMB	29.66	29.66	29.66	29.66	29.19	29.64	25.64	29.64
Pl	69.45	161.46	29.42	29.40	29.46	29.44	27.1/2	29.40
Tl	1 1	71.	71.	71.	77.	73•	73.	73.
TF (Fuel Temp)	78.	90.	85.	56.	99.	33.	90.	71 •
AL ARMA			45,	932				
IAB SG	. 8/7	77 ·@	0		• 9/9	15 0 6	0 0	
Р3	377	426	482	565	395	455	502.	561
Т3	743	787	¥33	914	767.	3/3	959	122
RES. VALUE	4.9424	4.9424	4.9424	4-9424	4.6928	4.6923	4.6928	4.6928
J ECX TEMP (JBT)	88	90.	92.	96.	90.	90.	96.	100
FiiO Obs	8570	10090	1/580	13810	9060	10500	12100	13500
FIC Thrust	8767	10346	11910		9213	10776	12449	14734
FNT	8878	10451	12021			10387	1	
FNI 77 <sup>0</sup>	8900	15453	17051			10914	1	I
NLO Cha	73-73	7725	:169	8503	7574	·	8380	
inc	7271	7684	8063		7429	7850	8255	5702
INIO Obs	11630	11920	12225	12727		12051	1	T
- MIC acm	1-179	11765	17069			11901	7	
WFO Cbs				9110		6676		
WFC fiel flow		6375			Į.	6370		
WFC 770								
P5.1 Obs	24.0	28.7	33.5	40.7	25.9	30.9	35.5	110.6
P5C	54.41	59.86	54.75		56.70	62.16	67.05	72.41
E.P.R.	1.635	2001	2.177	•	1	2.077		
DELTA P Cos	41.6	49.6	56.6	1 .	i	52.0		65.0
WAIC	112.4	325.1	2393			731.0	7	258.4
T5 Obs	852	910	967		1 — ———	939	990	10:4
T5C 77°	139	727	954		890	451	1003	1068
TLQSC	1753		1966		1797	1901	2016	2164
TLQS (A)	1728	1639	· · · · · ·			1852	2008	2144
START TIME:	STOP	TIME:		TO	TAL RUN		hrs.	min.

IDIE TIME / hour 20 m/vites

Time AT 70% 20 Additional missies

TF-41 PERFORMANCE CALCULA	TION RECO	)RD	<del></del>	TSH	11393	TYPE	K 120分 1EST デス	,		
		<del></del>			797		PLATE			
14/		SEQUE	NCE NUME	BER		DATE 5	CXT 3	5		
THET EXAL NUMBER TO	ں	OPER/	itor Co	arpente	7,7	ins Pecto	ارس آرا	use)		
			TCRED/RE			12	T 2 230 1 1	6100		
DATA	Eniv.	2 nd 2.	12mivaro	011.0	304 U, 5	6	7	8		
DEM POINT	53	53	53	53	55	55	55	55		
PAMB	79.79	29.79	29. 79	29.79	29.70	79.70	29.70	24.20		
P]	29.60.	29.57	29.55	29.52.	29.54	29.52	29.49	29.47		
71	85.	25 Se •	85.	86.	90.	90.	91.	91.		
TF (Fuel Temp)	84.	86.	89.	90.	38.	59.	92.	92.		
ARRA IIA .			45.	937						
IAB SG	. 8239a ° .8252 a °									
Р3	383	427	475	530	387	435	475	545		
Т3	775	813	857	912	792	5 34	871	938		
RES. VALUE				/	4,445	4.405	4445	4,444		
j box temp (jat)	./	•	8	•	110.	112.	115.	116.		
FNO Obs	2700	10050	11350	12900	8800	10250	11380	13270		
FIR Threat	8894	10248	11598	13216	8954	10465	11644	135.63		
FNT	8930	32801				105%	<del></del>			
FKT 77°	8982	10384	1			1323	-	13708		
MLO Obs	7490	7855	8210		<del></del>	<del> </del>	·	<del>,}</del>		
NLC		7351		2423			2996			
MIO Obs	11225	12070		ł.	1	1	1245	112820		
NHC /7 e3m1	Y		12020			·		-		
. WFO Cos	1		I					8910		
WEC Jul flow	5426			1	1		-	5877		
WFC 77°						-				
P5.1 Cus	124.4	28.6	32.7	37.7	24.8	28.9	32.7	32.7		
P5C	1				}			20.30		
E.P.R.			2.137		·	<del></del>	·   · · · · · · · ·			
DELTA P Cos	42.3		l	I	t	i				
WAIC						_	the same of the same of	755.4		
T5 Obs	1		1120	1	•	1	1	1		
750 77°		1045				419	+	· <del> </del>		
Tl;QSC	, , , ,	1866			<del> </del>	185-7	+	<del> </del>		
TLQS (A)	1795	1900		37/3	1843	<del>                                     </del>	<del> </del>			
START TIME:		TIME:			TAL RUM		hrs.	min.		

F-41 PERFORMANCE CALCUL	ATION RECO	RD			SH 2 SO	14 14 a 59 l	11		2 77 23
ENGINE SERIAL NUMBER /	4/234	SEQU	ENCE NUM	BER			DATE.		
TEST CELL NUMBER		OPER	TOR				LISPEC	TOR	
	DATA	TO BE	STCRED/R	COR	DED			***************************************	
DATA	5ml V	5M/N	111110	アバ	1	57110	10011.0	18002	611111
DEW POINT	55	55	55	5	K	56	56	56	56
PAPB	29.56	29.56	29.5%	29	56		79.5	7 27.57	27.57
Pl	29.40				1		1		29.35.
Tl	39.	69.	69.	69		69.	59.	69.	69.
TF (Fuel Temp)	59.	10.	92.	94		90.	10.	92.	75.
AL ARFA			4	5,	tre				<u> </u>
IAB SG	.82	47a 60	<b>つ</b> 。			• <u>ý</u> ?	37 0	€O°	
73	382	430	480	5	3/5	385	425		525
т3	745	787	634	83	4	735	784	839	877
RES. VALUE					I	1	-	1	
J BCX TEMP (JBT)		•	/		1	1			>
7110 Obs	8560	10000	11410	12	950	8390	19950	0 11580	12930
FNC Chrunt	8784	10289	11770	13.	390	8603	1073	0 11940	13356
FMT	8890	10399	11881	13	101	8713	1030	1/ 12350	13467
FNT 77°	1917	10426	11913	13	31	8:735	1036	7 12081	13501
NLO CDs	7400	7?60	8145	89	164	7325	223	5 8169	8500
нс	73/5	:499	8051	ξ,	36	7246	764	6 5075	8402
іщо Сра	11745	11980	12779	12	ŝ <i>31</i>	4673	1193	8 12754	12515
MIC RIMI	11610	!/843	12150	17	જ	11538	1180	0 /2/13	12371
, WFO Cos	5385	5255	7365	85	50	5220	625	0 7486	8433
WFC Luel How	5532	13499	7665	5	20	5400	648	5 7779	8765
WFC 970	-7	2	1	2705				<u> </u>	=
P5.1 Cus	24.5	28.6	37.5	35	2.	24.7	28.	7 33.9	37.9.
PSC	55.48	59.55	35.05	76	.67	55.65	59.9	2 6545	69.70
E.P.R.	1		,	1			1	7 7.187	
DELTA P Cbs		17.8	55.2					+ 55.6	
WAIC	209.4	723.0	237.0	221	7			2 2327	
T5 Cbs		1022			18		1	4 1098	1148
т5С 77 <sup>0</sup>		10:13	1115	T	10	978	100	51120	1171
ThQSC	17.86	1683	2006	2	170	1769	185	4 5016	2115
ThQS (A)	1252	1848	1969	20	28	1735			7078
START TIME:	STOP	rihe:			TO	TAL RUN	TIVE:	hrs.	min.

IDIETINE / hour

Additional Ministry That 15 min

	4.	4373						
TF-41 PERFORMANCE CALCULA	<u>S.</u>	4373		FRC CK	1434	) T5	CX //s'.	<u> </u>
IF-41 PERFORMANCE CALCULA	TON REC	JKU		750			A PLATE	
ENGINE SERIAL NUMBER /4/	494	SEQUE	ENCE NUM	EER 797	<u>,                                     </u>	DATE /	7 SEPT	5.7
TEST CELL NUMBER 2		CPER	VICR_			i is pect	R	
	DATA	2 35 OT						
DATA	Junia 1	57/11/2	9/13/2	61111	550	6	7-11-1	3.01,0
TRIOT WEG	18	48	40	48	45	45	45	45
FAMB	29.71	29.71	29.71	Z9.71	29.68	27.68	24 .63	24.68
F)	24.54	29.50	29.78	29.46.	29.50	29.4	8 29.46	29.42
71	58.6	59.1	59.6	61.4	65.1	65.5	65.4	65.5
TF (Fuel Temp)	₹3.	89.	91.	92.	87.	90.	92.	94.
. Al. ARRA			<i>4</i> 5.8	23	· · · · · · · · · · · · · · · · · · ·			
IAB SG	.8/2	25.3	60°	,	• . 8.	1850	650	
P3	382	428		555	379	430	475	553
T3	722	764	804	883	734	778	817	890
RES. VALUE	سد اداه	a singuat	2000	ري .	5.1873	5.187	3.1813	5. 7372
J BCX TEMP (JBT)	\$ 44.00	in rates a	19416 TIDE		53.	90.	94.	95.
710 Obs	8760	10010	11530	13700	8700	10200	1/530	13620
FIR Thurst	8458	10264	11830	14120	8901	10459	11848	14047
FAT	4064	10375	11960	19231	9011	10570	11954	14/58
FNT 77°	9092	10 "=)	11990	14266	9034	10596	11989	14193
NLO Chs	7.344	7716	8066	8633	7370	17.74	4095	5 687
nic	7336	7704	६०५५	8655	7317	7716	8035	5622
1710 Obs	1		ł	1		1	12173	1
MIC [3.0,1]	11311	11894	12029	13541	11603	11833	12083	12522
WFO Obs	5259	6214	7/23	8444	5292	6272	71413	5579
WFC gleer	5455	6438	7355	9281	5442	6492	7405	9225
MEC 778 manus Entra manage			1 2 m 3 m 1 m	19.00, 19475-	1106 F ( )			• • • • • • • • • • • • • • • • • • • •
P5.1 Cts	2-7 . 8	29.7	33.4	39.9	24.8	27.2	33.1	39.7
P5C	55.2	30.3	64.5	7/07	55.7	60.3	64.5	71.5
E.P.R.	1.8%	2.01	2.16	2.31	1.56	2.01	2.15	2.39
DELTA P Cbs	12.4	41.1.2	55.7	66.5	42.2	41.4	55.8	66.0
WAIC	211.8	225.4	257.5	255 .6	211.5	225.	7 237 • 5	254.9
T5 Cbs	92.3	987	1043	1134	843	899	948	1020
T5C 77°	960	1033	1057	1178	872	928	978	1051
TLQSC	2771	1:0-17	1757	2148	1774	1883	1967	2/39
Tl:QS (A)	1574	1801	1885	2074	1725	1533	1115	2583
START TIME:	STOP	TIME:		70	TAL RUN	TIME:	hrs.	min.

I.Dle Time 40 Min Additional military Penen Char - 25 min

F-41 PERFORMANCE CALCUL	ATION REC	ORD		FAC CX TSA 3	14395 13.26.	TYPE	X // 85 TEST = /A PLATE -2	<del>,</del>
FROIDS SERIAL NUMBER /	41494	SEQU	ENCE NUM	er 7		DATE		<del>-                                    </del>
TEST CELL NIPHER	2	OFER	ra			i is fecto	R Bord	
	DATA	TO BE	STCRED/RI	CORDED		,	in all	
DATA	Grand 1	7.71.0	3	7-14-2	5711.57	1:, 11:12	Naus y	3 -16.30
DEW POINT	45	45	45	45	42	42	42	42
PAMB	29.66	29.66	z9.66	29.66	29 .60	29.60	29.60	1
רַץ	ì	1	ì		l .	i	19.32	
Tl	69.3	67.2	69.7	69.2	75 •	75.	75.	75.
T? (Puel Temp)	93.	92.	95.	96.		90.	95.	97.
AL AREA			4	5,8	23	<u> </u>		
IAB S3	. 3	6 وكرة ا	0 0		• 67	દા હ	60	
Р3	383,	447	479	545	330	923	470	535
т3	744	789	829	891	751	785	832	886
RES. VALUE	5.4714	6.474	64714	64714	5.63	5.63	563	5.63
J ROX TEMP (JET)	90.	93.	97.	98.	95.	76.	99 •	10! •
FNC Obs	६७८०	10470	11650	134∞	8750	10020	11515	13130
FNC Thuce for	1 8357	1075-1-	11903	क्रमध्यक्ष ्र	8971	10294	11869	1357
FICT	9067	10862	12094		90%1	1040%	11980	13689
FNT 77°	7090	10889	12120		9104	10435	12010	13719
MLO Cha	7419	7835	8163	5883	7430	7755	8154	860
KLC	7337	7764	8069		7301	7629	8022	846
MIO Cos	11743	11992	12241	12618	11779	11960	12241	1259
1310 13/19/1	11613	11883	12101		11544	11765	12042	1234,0
WFO Cbs	5373	6467	7342	3774	5340	6150	7290	850
WFC free!	5521	4683	7570		5484	6318	7494	8812
WFC 77°	\ <u>`</u> \	$\times$	X	>/.	><	5,-	X	X
P5.1 Cos	24.8	30,0	33.5	39.1	24.6	28.4	33.1	38 . 3
P5C	55.7	61.2	64.9		55.61	59.65	64.81	70 · 2
E.P.R.	1.86	2.04	2.17	•	1.858	1.993	2.166	3.34
DELTA P Gos	142.4	50.2	51,04	55.3	92.2	48.	55.6	63.8
MVIC	211 .9	227.2	238.7		211.7	2,23.2	237.6	251 •
T5 Obs	1872	932	978	1043	870	713	969	1022
T5C 77°	592	957	998		877	920	976	1035
Those	1779	2250	1986		1772	1850	1984-	૨૦%⁻
Thos (A)	1147	2201	1953		1764	1441	1975	207

JAME Time 2 hour 15 min Additional milling our 25 mile

					FNC CX 1-139-2 T5 CX 1/17 T5N 2826.5 TYPE TEST 1/2 TSO 6 CC. ( DATA PLATE )			
91494	SEQU	ENCE NUM	esr "/	OR TYGTY DATE 10 COUT				
	OPER	ردنے TOR	1 ( 11)	<u>.</u>	I:SFE	CTCR	11:01	7.79
31112	371110	13.00	4 14 3	5	6	5	7	8
45	45	45	4.5					
29 .58	29.58	<u> </u>	29 · 5%	•	.		• .	•
29 -40	29.38	29.34	29.32	<u>.</u>			• .	
72.	72.	72.	72.	•			•	•
13·	9d.	95.	97.	•		[	•	
			t5.8	23				
.92	11 6 6	C		•	(	6	o °	
373	430	475	540					
738	784	828	989					
5.964	339.5	5.968	5.968					
94.	96.	97.	100.	•			•	•
14530	10030	11530	13290					
	1		1					
1		1						
		1	i					
-1			1					
	(							
11536	11786	12067	12429					
5166	6209	7250	8660					
5317	6408	7494	3965					
13.7	28 • 8	33 • 2	39.0					•
51.66	60.03	64.76	7092	•		•	•	
1.827	2.006	2.164	1.370		.	]	•	
4.	117. 4-	15.8	14.6			•	•	
209.2	224.0	237.9	253.0	•			•	
४५४	9110	1771	1034					
871	930	985	1048					
1751			1					
1730	1	1951	3.0%6					
	11494  DATE  5 1 58  29 58  29 58  29 58  29 59  313  738  5968  94 6  94 6  7235  11695  11536  5166  5317   237  5166  1.827  416  209 22  858  471  1751	1494   SEQUE   DATA TO BE SEQUE   STILL STEEL STILL STEEL STILL STEEL	1994   SEQUENCE NOTE   OFERATOR 273   OFERATOR 27	TION RECORD    1499   SEQUENCE NUMBER   7	TISK 2836.5  TEST	ATION RECORD  ATION RECORD  ATION RECORD  ATION RECORD  ATION RECORD  ATION RECORD  ATION RECORD  ATION RECORD  ATION RECORD  ATION RECORD  ATION RECORD  ATION RECORD  ATION RECORD  ATION RECORD  ATION RECORD  ATION RECORD  ATION RECORD  ATION RECORD  BE STORED/RECORDED  ATION RECORD  ATION RECO	TION RECORD  TSS 3886.5   NFE   11494   SEQUENCE KINGER   797   DATE   10   11494   SEQUENCE KINGER   797   DATE   10   11596   1399   1600   11596   1399   1499   1596   1498   1399   1499   1596   1498   1399   1498   1399   1499   1596   1498   1399   1498   1388   1499   1596   1498   1498   1499   1596   1498   1498   1499   1596   1498   1499   1596   1498   1499   1596   1498   1499   1596   1498   1499   1596   1498   1499   1596   1498   1499   1596   1499   1596   1596   1499   1596	TSI   33.6.5     FORE TEST   7.77

the policy

To Nation

C. Carrent

10 mg 3

TF-41 PERFORMANCE CALCULA	TION RECO	) RD		754 3	1436; 539 2531		1190 1887 F / PLATE	₹
ENGINE SERIAL MUMBER 12	1-177	SEQUE	NOE NUME	ER 79	ا ما	DATE :		7.5
TEST CELL NUMBER II.		OPER A	itan Ac	or Lar		NS PECTO	3	tnice.
		TO BE S						
DATA	1 1	עייייי	/メペルン 3	7/11/1	10000	14131	7	B
DEW POINT	50.9	50.9	50.9	50.9	56	56	56	36
PAMB	29.80	ეე • გა	अ•६०	29.50	30 · ( u	29.54	29 . Fil	29.84
P]		29.58			ŀ		I .	i
Tl	68.3	67.5	68.0	0.39	62.	12.	82.	(2.
TF (Fuel Temp)	86.0	6.83°	90.0	90.	70.	92.	<i>i</i> 3.	44.
. Ali AREA	40	5.617	7					
IAB SG	0.82	<u> </u>	0		. 37	11 0 61	() °	
F3	378	425	473	530	383	475	485	70.2
т3	735	774	990	868	770	507	860	911
RES. VALUE					4.794	1.7961	4.704	1. 794
J PCX TEMP (JBT)			•		104.	10%.	115.	14.
FNO Obs	8540	10010	11440	13050	8350	9754	11685	1.210
FMC Thurst	9697	10226	11764	13393	8668	13167	11933	
FNI	17809	10337	<b>\</b>	13504				
FHT 77°	8830	10362	_	13533		1	17576	
NLO 065	7325	7695	3085	4495	7453	7788	8240	5636
NLC	1248	762.0	8007	84.08	72 13	1.50	8051	
MHO Cbs	11646	1459			11632	17057	12503	1750
MIC CANTON	•	11744	1	)	1	ı	4	
. WFO Cbs		(150			5413	1749		
WFC Luck	1	6354	3	1	5457	5311	16 30	
wfc 77 <sup>6</sup>				<u> </u>		-		
P5.1 Cbs	24.5	28.7	33.6	38.2	24.4	179.0	34.0	34.0
P50	55 • 31	59.77	J. 40.	67.80	55.61	59 68	\$6.20	•
E.P.R.	1.848	1.997	2.169	<b>২</b> • 332	1.861	1.994	2151	
DELTA P Cos	i	47.6	1				56.0	
WAIC		221.9				~~~~~~~	~~~~~	
T5 Obs	955	1013	ŧ	1135	ł	1	Gar	
T5C 77°	476	1037	1099	7	677	7	7.7	
TliQSC	1769	1873	1979	7-	1	100	20,00	
This (A)	1733	1830	·	ર તું કહે	-,	<del></del>	<del></del>	
START TIME: 0630		TIM:		W.T.	TAL PIPE		hrs.	nin.

IDIE Time 40 miles Additional not Hand Time 15 min

TF-WI PERFORMANCE CALCULAT	NOSR KOLT	RD	<u>.</u>	134	14899	TYP	TEST ;	112
DUATED AND THE MILITIAN			***************************************	TSO	1		PLATE \$ 7	
ENGINE SERIAL NUMBER 14	1357		<del></del>	BER 50			SEPT 8	5
TEST CELL NUMBER 2	er meneral experience	OPER!				INSPECTO	R	
			STORED/RE		380.02	5000	13711	12101 100
DATA	5.7.	2		6-11	<del></del>	6	7	3
DFW POINT	47	<i>47</i>	47.	47	46	416	46	46
PAMS	29.70	29.70	29.70	29.70	29.66	29.66	29.66	29.66
Pì .	29.52	29.48.	27.76	29.46	27.47	29.43	29.44	•38
ำา	7/•	72•	720	72.	75.	75.	77.	76.
TF (Fuel Temp)	73.	80.	٤٤,	59.	80.	58.	90.	91.
. AL AREA			45	१५०				
IAB SG	. 8/92 @ 7c				. 8%	5/ @	79°	
Р3	386	430	490	538	3 ? 6	456	480	228
т3	750	785 .	દ35	८६२	733	802	842	913
RES. VALUE U	Theirings	ور نسخت الله	and with the	الحرث ومنسد	48745	787-1	7.8/4/2	48143
J POX TEMP (JBT)	* ** ** <b>*</b> ***		s Geografia	- 41		79.	102.	111.
7NO Obs	8750	10030	11630	19100	2660	10270	11650	13710
FIR Thrust	8940	10275	11945	15557	8857	10530	11863	14141
FNT	9050	10,385	12053	13888	<i>३</i> ५५ <i></i>	10640	12014	14252
FNT 77°	9073	10411	12083	15702	5990	10067	12104	14288
MLO Cbs	74.13	7750	8144	8222	7415	7838	8/83	8756
nice	7321	7644	8032	8438	7273	7709	5034	8604
liilo Cba	1/257	11931	12.247	12563	11775	12041	12255	12712
MHC /L/2/11	11607	11167	12079	12391	11542	11843	12061	1249
. WFO CUS	.5375	6253	7379	8572	5385	6436	7925	9056
WFC Luci	5579	64.53	7613	8.807	555/	15544	7653	9381
WFC 77° /					- 1	1		
P5.1 Cos	20.2	28.7	1	ì	24.5	29.2	33.3	39.8
PSC	55.70	59-8	64.9	69.6	55.9	60.3	44.7	71.7
E.P.R.	1.04	2100	2.11	232	1.85	201	2.16	2.39
DELTA P Cos	92.5	49.2	36.0	163.9	41.7	1/7.2	. 5K • K	65.9
WAIC	211 .8	223 •3	237.8	250 25	210.5	1225 . 5	257.5	250.8
TS Obs	971	1024	1039	1152	172	935	984	1056
τ5C <b>7</b> 7°	286	1037	110%	1166	279	942	986	1060
Thusc	1787	1871	1993	2093	1775	1897	1995	2146
Thus (A)	1761	1849	1	2069		1588	1475	21:11
START TIME:	STOP	TIME:		TO	FAL EUN	TIME:	hrs.	min.

IDIE TIME I hour field thank nellowy Time 20 min

F-41 PERFORMANCE CALC	JEATION RECO	טאט	·	139 130	1201-1 3 4080	DATA	PLATE	
ENGINE SERIAL NUMBER	191551	SEQUI	ENCE NUM	er 79	3	DATE 20	AUG	85
TEST CELL NUMBER 📿		OPER!	OPERATOR ADD MASSAGE INSTRUCTION AND PLANS					
			STCRED, 'RI			Z.O.		
DATA	1 1	4/11/.0.	9747.L	6.117 U.	5.1.5	37.8	7	F11/11/
REW POINT	58	58	S S :	58	56	56	56	56
FAME	રવ ∙6%	રુવ •64	39 - 64	29 .64	29.70	29.70	29.70	39.71
P1	29 -51	29.49	29.47	29.46	<u> 29 •53</u>	G.51	29.49	29.46
กา	1	75•	l	1	į	h.	71 •	'O •
TF (Fuel Temp)	36.	90.	91.	92.	97.	93.	લુવ.	95.
All AREA			45	.513				
LAB SG	. 82	22:3 G	) °		. ? )	3106	0	
Р3	378	428	478	548	380	430	484	574
73	733	794	83%	905	741	873	851	9/9
RES, VALUE					4.7869	4.79.99	4.7899	4.789
J BCK TDMP (JBT)		•			100 •	101 -	104 •	105.
FNO Cos	9560	10010	11500	13430	1510	10090	11530	14160
750	18744	[	11003	1				1458
FMT	1455		11914	1	1		11942	
FNT 77°		]	11944	Ţ <b>,</b>		i	11972	14725
NEC Cos	7273		3135				8107	8794
:IT.S	7243	7632	7994	8536	7251	7626	799%	8676
MIO Cos	11778	i	12390	i	۱.	1	ไกวรว	
NHC	111554	1	1	12497				
, WFO Obs	5263	6233	7287	9867	5226		7219	9333
WEC	5285	6391	7479	9139	535%	63%7	7540	9647
WFC 77 <sup>0</sup>	4		-		-			
P5.1 Cbs	124.0	રુજ ∙ ક	33.2	39.6	29.1	28.8	33.4	41.4
P50	54.87	59 • 60	164.57	71.32	51.96	59.90	64 -77	73.20
E.P.R.	1.833		ļ	;	į .	1	1	1
DZLTA P Cos	ł		55,0	1	1	1 .	1	1
WAIC	209.6	, .,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,		1	-			1321
T5 Obs	074.	1034-	1093	1	854	979.	967	10/52
T50 77°	977	1039	1099	1	1	1929	983	10-3
These	17/49	1875		2150		1968	1979	2181
ThQS (A)	1794	1866		a 130	1735	1946	195a	2151

I Die Time Thous 10 ann

PHILIPPING AND INTO THE DUST PROMIT

TF-41 PERFORMANCE CALC	ULATION RECO	RD		FAC CK				
ENGINE SERIAL APBER	1411511	SEQU	ENCE NUM	<del> </del>	<		·····	· <
	-W1) _		NTOR 6	<del>,</del> ,	i	IISFECTO	AUGA Ros	
4 4 7 3 C Tr Grade (1997) (2007) (2007) (2007) (2007) (2007) (2007) (2007) (2007) (2007) (2007) (2007) (2007)			STCRED/RI				7.0.5	<u> </u>
DATA	7-010	7011N 2	Bond 3	アルル	5000	6.00	7	6-11-10
DEM POINT	58	58	58	58	59	59	59	59
FAVB	<del></del>		29.73		<del>i</del>	<del>                                     </del>	29.7/	24.71
.b.)	29.55		<del></del>			<del></del>	<del></del>	-
71			:			92.	92.	93.
TF (Fuel Temp)				·	· · · · · · · · · · · · · · · · · · ·	26.	18.	48.
, AL ARRA		5-89			<del></del>	45,	298	<u> </u>
IAB SG		q & G		: • }	. 92	3/30/,		
P3			462	500		1395	435	476
т3	774	穷13	846	i	761	795	831	865
RES, VALUE					1	6,250	6.250	7
J BCK TEMP (JBT)	7.	/.\	/.\	/.\	l .	1/2		116
FNO Obs	8700	10000	1/200	12400	·····	T	10250	11550
<del>earais aireana aireana airean</del> PNO	级62			TENEFATERS	TALL COLUMN	9327	10 575	PROPERTY NAME OF THE OWNER,
FHT			11565	,		<del></del>	والمعطوبة والمستشمرة	
77°	2445		,					\
NLO Cbs	7490							
HLC	72.72		1000	, , ,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,	,	17400	77/9	7993
MIO Cbs			12375	i	i	1	12290	12460
IRIO	1/632	11345	1	12193	TANKE MAN THE	11695	1/389	12042
WFO Ges	5940	6370	7100	3000		_	6648	7490
WFC	5467	6336	7/37	8045	5677	5772	6650	7506
WFC 77°						\		
P5.1 Obs	34.5	18.7	32.2	36.0	23.0	36.2	37.2	33.8
P5C		59.75		67.46		A	61.32	16c.11
E.P.R.	1.849	1.497	2.120	1				j. 2 - Luki
DDITA P Cec	91.7		53.5	1		44.4	51.	56.
WAIC	2/5.5]	23-3-1	233.4	243.9	204.6	216.1	227.0	238.1
TS Obs	996		1053		293	930	977	1023
75C 77°	469	1017	1062	ے در جبر <del>ہ کی آجا '</del> دھوجت ا	854	1998	943	996
Thase	1270	1859			1729	1802	1896	1970
Thes (A)	1816	1916	·	0.032	<del></del>	1865	1962	2042
START TIME:	stop			C 100 Sec Decision	TAT, PER	· 大学· · · · · · · · · · · · · · · · · ·	hrs.	lee==reec.=r: Filip •

TF-41 PERFORMANCE CALCU	LATION RECO	RD		F::0 CK 73:	164773 1265	T	S CK 11 G	7 - - (vi) f <sub>0 =</sub>
EMGINE SERIAL NUVEER /c	11617	SEOU	ENCE NUM			DATE	6 Aug	00
TEST CELL NUMBER #F		OPERA	ATOR So	: lache	٦٠,	 Iseo	oran Bor	ucued
	DATA		STORED/RI			·		
DATA	アバル	हराग्य 2	141111,00	31111	6000	70.6	720 7	741-15 0
FFW FOINT	SZ	52	52	.32	5!	51	51	51
PANE	29.69	2968	29.67	79.68	39.68	29.0	8 29.68	-,
P1	2952	29.49.	29.45	29.44	29.52	29.5	10 29 47	30,000
71	65.	65.	64.	64.	52.	62.	53.	33.
TF (Pool Tomp)	129.	92.	95.	27.	95.	76.	77.	10%
, AL AREA				45.	669			
1A3 S9	. 5,2)	1 360	o °	·	. 32	?9, G	c	
93	379	435	482	543	35	6-4	0 484	1550
TO	728	776	917	972	724	77	4 815	873
RES. WALK					5.33-3	14.33	43 5-3343	5-3243
J BOIL TEMP (JET)		<u></u>	/.	8	£7 <b>.</b>	37.	71.	95.
FMO Obs	9600	10550	11670	13270	8,500	11030	1750	ردين 3 /
PNO	18773	10406	11995	13677	9796	1560	17 17075	13:05
FMT							4 12185	
FMT 77°	2024	1543	12139	3.72	8929	57	1 12216	12,450
NLO Obs	7320	7760	3115	1545	7315	777	C 2150	13536
MIC	7265	7702	506 2	8419	7287	ب ربر	3 50-75	3 939
13i0 Cb3	11605	11900	12175	125%	11590	11819	10 12170	1255.
MIC	1/3/8	11211	12096	17453	11537	11. 1	1. 12168	12051
WTO CUS	SEALCH SELECT PARTY PARTY SALES	are the service of	7795	2595	5245	539	0 7350	1 (1)
WFC	15405		<del> </del>	·	<del></del>	·	1 7641	3/058
WFG 77 <sup>3</sup>	<u> </u>		3.5.12		1.530 E	سربهدن سریخ	- 12.00 E	
25.1 Gts	24.0	29.0	33.7	35.6	24.2	25.	5 38.7	35.7
P50	54.88	60.15	1,04.64	3 36	55.60	30 ,	(7 65. 1	2 70 .50
S.P.B.	1.5'34	7010	2.100	2.350	1.541	2.0	27 2,176	2. 561
DELTA P Coo	41.4	49.2	156.4	164.6	42.0	1.	5 - 7	155.1
WAIC	209.8	275.4	238.7	252.7	211.1	27:	. 4 73	2 (360)
TS Cas	1945	1009	11058	1123	841	300	1 - 7	10/7
T50 77°	974	15-15	1093	11:12	877	1-7.5	988	1000
THOSO	1765	1800	1986	: 113	1751	159	7 147	7 77.
Thus (A)	1715	1855	1126	7676	1595	1188	1 416	
FIAR THE:	Istor	TIME	PA 81-1-1-1 - 201 - 1-1-1-1 -	170	TAL RUU	TTT.	hro.	nin. markilin

TF-WI PERSORNANCE CÂ	LCULATION RECOR	10		738 7	C CX 14743 T5 CK 1  R 77 3-1 TYPE TEST  C Z C C DATA PLATE				
RHOIME SERIAL NUMBER	141630	SEQUE	ence nate	®R 29°	5	DATE	24	1 1. v., S	(3)
TEST CELL NUMBER	互	OPERI	via //	1 1/2		I MS PE	ota		
			STCRED/N		,	VEID !	<i>′</i> .		
and the second s	5-n20 13	2	977.0	مرز برد اذ	5/10/2	1.501.7		3412	37.7.0
DEW POINT	- 53-	5.3.	53	5-33-	-5.7	5	7	57	57
EANGE .	1-11 29.67 2	9.68	29.35	29 <i>L</i> 8	<i>29</i> °.69	79.	69	39.59	79.61
Pl	19.50				,				
Tl	71	??. <del></del>	13.	75	780.	15.		71.	7.1.
TF (Fuel Temp)	1.1. 8	द्र	35	16.	37: "	88.		44.	11.
Al. ARGA			4115	6,664		4			
MAD SG	. زن د	/5 B &C	, c		•52	<i>3</i> 2 3	60	)°	
P3	580	434	484	563	380	43	5	499	545
73	744 -	792	375	9/.L.	7.5%	30	.,	565	145
RES. VALUE					1	1		52662	52662
J BOX TEMP (JET)		0		-	1	104		109	11/.
FMO Cos	35:01	10030	11530	13620	8500	150	Yá	11240	1455
PSC	5710		1	!	1	1 '	-		
FNI	5825		1	1	i				
F:77 77°	1643								
NLO Ots	1385		1	1	1			1	5769
MC				9648	,	,		8/63	<u></u>
1310 Obs	11 701 !			,		-;	19	12,400	1289
BRO	1547	ALANG SERVICE		And it? Miletanders has	CAMPAN PARTERIES		NAME OF TAXABLE PARTY.	A MANAGEMENT AND A SECOND	
, W70 Obs	(21)				1			1	7340
NTO	1375	And the second of the second of	**	e was the to the distribution to the distribution of the distribut	CALL STREET		PF 2.5	STATES OF STREET	-
WFC 77°			أحد سمنة						
P5.1 Cbs	10-100	2807	5506	140.0	14.0	: É	5	34.7	-1103
₽≶C	14.89		1	1		1		A MARKET OF SHAPE	<del></del>
E.P.R.	1.73/ 2			i	i			1	·
DELTA P Cls	40.4		]					1	
WAIC		*******	i andre i dei den den den i i i i i i i i i i i i i i i i i i i	***********	~ · ************	+	-	Color Street Street, Colors,	I Herman gare - Marter in
T5 Obs	75.1.		1		,			768	·
T50 '77°	7. 6 h	~~~ ~~~ ~~ ·	, <del></del>	, <del>,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,</del>				75.7	
ThQSO				7145	<del>}</del> -				
TIÓS (A)				7 / 24				2.454	
STAPE TIE:		TWE:		70	7 -4 1 <del> 1 4 4</del>			hrs.	nin.

T-41 PERSORMANCE CALC	HEATIOH REC	080	1 PERFORMANCE CALCULATION RECORD					TS CK TYPE TEST FIR			
		-		120 _	185	<del>,</del>	PLATE				
RINGINE SERIAL NUMBER	14195					DATE 2					
TEST CELL NUMBER	2_	OFER.	ATCR (	12/15/h	79 <u>.C</u>	INSPECTO	B5-e1	S77.			
and the second section of the second		A TO EE			<del>,</del>	_,	<del>,</del>	<del>,</del>			
DATA	5112	3700	37.0	3 41.0	5	6	7	В			
DEM POINT	57		<del> </del>	57		ļ		ļ			
FAIR	29.70	29.70	29.70	29.70	•		•				
<u>51</u>	27.52	29.50	29.43	17.46							
Tì		73.	78.	18.	•		0				
TF (Fuol Temp)	12	194.	15.	l : i, L : io ・	•		.				
AUSA JA		·····		15,90	<u> </u>						
IAS SG	.5/	55 a 6	<i>o</i> °	!		9	0				
P3	372	405	435	470							
Т?	744	773	1	334							
RES. VALUE		1	\/								
J BOX TEMP (JBT)	7.\	X				Ι.					
FNO Obs	3510	19550	10470	1500							
PNO	8687	9408		1305							
श्तः .	5798	9918	10832	1	}						
РИТ 77°	4820	=1943	i	1 .							
MLO Che 🚜		7465	78/1/2	1		<del>-;</del>					
NIC	MANUAL PARTICIPANT SAUGES	7316	7709	7994	<del>'</del>						
MIO Cos				12209							
<del>- 1510</del>	B. C. A. B. J. MAAA. D. A. J. B. B. A. J.	AL PROPERTY.		1965	C. S						
WFO Obs	:	242	1	1 .							
and the second of the second o	سے در جوہوں ہے۔ اور اور اور اور اور اور اور اور اور اور	61/2	alls have no would also	40001. (4		-	AND THE SECTION AND THE SECTIO	- In-			
WFC 77°	><	><	X	15/	X	X	-><	1			
P5.1 `Cts	251.2	27.3	30 1	·							
P50	<del></del>			134.91	•						
E.P.A.				1. 169			1.				
DDITA P Cos				53.0		•					
ne dan mada senen West den senen Walio			7-15-5-1-	36.2							
TS One		1032	·								
150 77°	981		1067			İ					
Thqsc	1771		1907					<u> </u>			
Thes (A)	<del></del>	155)		1003							

INTE Time I have miditional miditional midition for the 10 min

F-41 PERFORMANCE CALC	ULATION RECO	P.D	*	180 C	SOCO STATE				
engine serial number	141952	SECU	ENCE NUM	35R 79		DATE 8	-13-5	35	
TEST CELL NUMBER	Ţ	CFER	ural Co	llecai		TNSPECTO	R Ross	,	
		TO BE	stoded/r	ಕರಡಾವಾ					
DATA	16 Min	بارزارين	13	7.11.27	5	6	7	8	
DEM POINT	58	58	59	58					
PAVB .	29.76	29.76	25.76	27.76	•	,	•		
P1	29.59	29.5	25.5%	29.53	<u> </u>		<u> </u>		
71	70.4	711.5	-1.7	72.6	•		-	•	
T7 (Fuol Temp)	9.7.	<u> 15.                                    </u>	107,	59.		<u> </u>		<u>  .                                   </u>	
AL ARRA	45	,900		<del></del>					
iab so	52	778 (	<u> </u>	<del></del>		4	С		
P3	365	<u> 106</u>	445	<u> 502</u>			1		
73	729	769	6:52	855	ļ				
RIS. VALIE		$\sum$							
J BOK TEMP (JBT)		<u> </u>			•			•	
FNO Obs	5400	9650	10730	1740				!	
FIIO	<u> २</u> ५५ ८	<u> 1865</u>	10-74	12717	 				
FNT	3673	9966	11035	12528					
7:T 77°	8495	9991	11113	12860					
NEO Ges	7290	7413	7:13	8355					
NLC	7195	7535	<u>  უგავ</u>	8229					
Mic Cbs	11565	11845	12097	12455					
Mic	11414	11678	11925	12247					
WFO CUS	5215	<u>රෙවුර</u>	16315	8150		*, ~W 71, 257 W 79 /			
WF 0	53106	5251	7000	8388	!				
<i>utc</i> 77°							-		
P5.3 Cbb	23.7	27.13	131.2	36.7					
FSC	54.50	<u>58, 8,</u>	162.36	1.8.13	•		0		
E.P.R.	1.521	1.965	7.024	2.279		•			
DELIM P Coo	40.2	11,00	51,	59.8				2	
WAIC	707.3	213.5	220.8	244.5					
75 Oho	1951	1019	10%	1135					
т50 <b>?</b> 7 <sup>0</sup>	7,77	1032	3 2 3	1147					
Those	17:37	1397	1966	2090					
T <sup>1</sup> /2S (A)	1760	1863	19/217	1,5069					

Flike Time 50 mil

PAS LENSON DON 1270212 LOEV, 12-77 LENSON LOEV, 12-77 LENSON DON TAME OF MAY

F-41 PERFORMANCE CALC	ULATION RECO	ORD	1	TON SISST TYPE TEST FIX					
REGINE SERIAL MUMBER	141952	SEOU	ENCE MUM		7	DATE 9	2.3 -	÷5	
TEST CELL NUMBER	-11_	OFER	vial 24	LNOLE	$-i\hat{\varphi}$	INSFECTO			
er i de l'annuel de l'annuel de l'annuel de l'annuel de l'annuel de l'annuel de l'annuel de l'annuel de l'annu	ערגם		รา(ลโรย/ณ			REChi	X CX U	1177	
PATA	\$ "and or	4241.20 2	3	50.0	9.7.0	7-11-6	7	8	
DEN POLUT	53	53	513	S-3.	53	53	53	53	
PANS 7	29.66	29 46	29.16	29.16	·° .64	24.63	24.68	2169	
P3		ì	1 .	•		10.46			
71	57.	39.	cy!	85.	5%.	55,	<i>9</i> 5.	<i>३</i> ५.	
TF (Fiel Tamp)	189.	<sup>7</sup> ي.	75	55.	ā.,	94	76.	77.	
AL ARRA			4.5	. 878					
IAB SG	.43	0706	0 3		.51	9/ e c	9°		
P3	370	405	445	5000	3%)	4-4	455	505.	
T3 <sup>*</sup>	769	807	441	892	795	314	843	-389	
RES, VALUE			/	<b>\</b>		\/\	/	×	
J BCX TEMP (JBT)	1	/-		7-	1	/.	<b>^.</b>		
FMC Cos	3390	9550	10680	12350	1198	10030	11100	12450	
	5758	9764	167/2	12689	4338	10266	11382		
FIT			1	12500		10376	11492		
FNT 77 <sup>0</sup>	4570	7599	11030	12832	9482	10402	115.21		
MLC Cbs	t		1 .	3488	1	7349		5474	
NIC	7/82	7531	7825	3245	2427	7645	79/9		
MHO Obs	11749	12000		1		12075		12612	
15:0			A	1	1	11766			
, NFO Cos						6472	;	4,260	
1/7/0	1			1	, , , , , , , , , , , , , , , , , , , ,	4534	4-4-0-15 ED-1-4-4		
WFC 77 <sup>9</sup>	1><	X	><		X	X	1	<i>&gt;</i>	
P5.1 Cbs	23.5	17.5	3/.0	36.4	1/1 · /	29.1	32.2	33.9	
PSC	54.33	58.54	32.25	67.78	57.07	1,0.24	23.54	,	
2.7.3.	1.8%	1,956	2.000	1.272	11.327	2.073	2.123		
DBLTA P Cbs	40.2	4/3.3	51.8	60.0	44.3.	18.4	53.2	(0-4)	
WAIC	207.6	219.7	230.5	245.2	205.7	233.9	253.0		
TS Con	1002	1059	1129	1119	104/1	1079	1119	1174	
₹50 77 <sup>0</sup>	1993	1029	157.12	1139	1:16	1058	1078		
These	1747	1867	1939	2060	1533	1000	1975		
Tips (A)			19.80				<del> </del>		

A Jille Time 1 hour 15 min

MAG LEGISFE (#0) 19700/14 (REV. 12-77) (FRONT)
Additionally Following Park Track 35 min

JLATION RES	ORD		761 3	185	TYPE TEST DATA PLATE				
141952	SEQUI	פונב ווְעוֹם	EER 799		DATE 27	50,00	:S		
•	OFER	ATCR			INSPECTOR				
			CCCDSD		**************************************				
5 171.1	2	19 m/N 3 4	7-71.1	5	6	7	8		
159	59	59	59						
	1	·	1	•	1.				
	[		[	1					
1	1	i .	1				,		
,		3	1'	•			1		
		1, 6	45.	8 88					
.52	023 (			L.	9	0			
1	1		547						
751	795	339	7/3						
	i	i							
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1	1	1	i						
and the first of t	. Afterna and an and and a design of	والمستعدمة فلاغدها	. Tarantal Carried State of Control of Control			E PARTITAL SALES			
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1		· · · · · · · · · · · · · · · · · · ·	1						
1	1	:	i						
7277	7651	8046	36/6						
	1	1	1						
11750	11737	12/05	17578						
5373	16291	77 61	0059		, TETRAL				
5430	6447	7462	9353						
				$\searrow$					
24.2	23.7	13.0	140.2						
55.11	57.95	64.49	72-13			•			
1.342	2.003	2.155	2.9/2 .						
47.0	149.9	5:05	1/2/00)		0				
211.6	275.3	1737-5	256.6		•	,			
375	1935	786	1065						
371	770	993	1012						
1222	1833	1225	2/22						
1223	1184	1969	2163	]					
	141952 2  DAT.  SMI  SMI  SMI  SMI  SMI  SMI  SMI  SM	EATA TO BE SMY 5 79 9 9 9 9 9 9 9 9 9 9 9 9 9 9 9 9 9	MIPS 2   SEQUENCE HUND   OFERATOR   OFERAT				TATION RECORD		

INTO TIME 2- hours nullifornit million Fine Double

ULATION REC	ORD.				TYPE TEST //K			
141952	SEQU	SEQUENCE NUMBER 799 DATE 26 SEPT						
	CFER.	CFERATOR Soryel INSTRUCTOR BUERNAM						
	TO BE							
5 41.0	5 19110	12.3:2	7.6.0	ψπυ Φ 5	الرياس ل	General	7.11.85	
56	56	56	:06	60	60	60	60	
29.67	29.67	29.67	27.67	29.63	70.68	27.63	j ———	
29.47	29-46	29.114	20.42	27.45	21.44	29.42	29.40	
68.	128.	68.	62.	71.	71.	71.	71.	
85.	90.	23.	95.	97.	93.	96.	59.	
45	.828							
0.72	020 6	o •	ì	. 53	20.43 6	0°		
374	4/9	467	5/7	372	1420	471	568	
733	775	816	866	739	732	826	ł	
		<u> </u>		4.866	4. 166	4.366	4.336	
	1	<u>&gt;</u>		1	•	1	105	
3570	10010	11430	13.790	1500	10000	11480	1423	
5775	10269	11752	13/31	5637	10264	11810	1469:	
15536	10350	11863	13212	18509	10374	11920	1430	
1	1	1			1	1		
,				1				
7267	7612	7997	3393	7201	7:73	3022	877	
1	i	1	i	1	;		1	
11453	11776	12075	12404	11461	11767	12070	126.86	
	1	1	1	. —				
5424	4265	7417	1559	5417	6371	7461	9725	
	$\geq \leq$		<u> </u>					
24.2	25.5	33.0	37.7	14.3	125.5	32.2	41.5	
55.16	59.68	64.40	69.43	55.25	59.72	64.75	73.5	
i	1	į		1	1		1	
111.8	118.2	5401	62.0	11.9	48.1	55.4	64.0	
211.0	223.6	236.1	12:2-6	211.4	223.6	237.3	252	
	}		i		1		i	
	1	1	1	1			1	
			1	1	1	T	-	
i	1		1		i		1	
	141952  DATE  DATE  5:41  5:41  5:45  29.67  29.67  29.49  73.2  35.70  57.75  53.56  35.06  35.06  1.893  73.45  72.50  57.75  53.16  1.893  11.60  95.41  17.66  1.776	DATA TO BE STATE OF THE STATE O			### SECURICE NUMBER 799    COMMINGER SECURICE NUMBER 799   COMMINGER SECURICE NUMBER 799   COMMINGER SECURICE NUMBER 799   COMMINGER SECURICE NUMBER 799   COMMINGER SECURICE NUMBER 799   SALV STALV 12 30 71			

TF-41 PERFORMÂNCE GALOU	HATION RECO	RD	; ;		(५३ <u>० =</u> (५३० =	115 CK		<del>,</del>		
RHOINS SERIAL MUMBER	14 > 5/2	SECTO	HCE MUYE		5/			<del></del>		
THAT CALL NUMBER			SEQUENCE NUMBER 79/ CATE 15 AUG 95  OPERATOR Soil - CHSTEOTOR BOX WILLIAM							
The order		MA TALIBAM SHAM	TCRED/F1	AND AND AND AND AND AND AND AND AND AND	( -		11 K. F.	<u>''                                   </u>		
DATA			101.10		5 5	7.W.J	7000 7	3.00		
DEM POINT +	55	55	55	55	55	55	55	55		
EXXZ .	29.5/	- <del></del> }			20.61			<del></del>		
P3 :	29.34									
T1		i	69 ·			47.	ì	67.		
TF (Fuel Temp)					90 .	1		75.		
AL ARTA I			75	7-72			L'			
IMB SO	-5(23	10.6				<;	o °			
73	375			562	375		ī	568		
12	739				i .	783	828	912		
RES. VALUE	4,002			4002-		·		4,:02		
J BOX TEMP (JET)	90.	93.	76.		89.	1	-	97.		
FNO Cos	4560			1	}		11560	14019		
PNC				1:37%		1	1	1		
FUI				성무성의		,				
FMT 77°	3935			<u> </u>	i	1	1			
NEO Ohs	735%	7796	7/66	3779	7525	7777	8/47	\$78%		
red ,	72.7.8				1		1	8704		
NHO - Cha	1154			;		1		1267		
тно	111128	11804	12037	12518	11512	1:720	12067	1055		
WFO Cos	5212	4312	7245	9109	5162	6268	73/1	19210		
भारत है	594.1	655h	7549	9553	5365	13523	7622	1965		
٠٠٠٠ رود ١٣٠٥	><	$\geq \leq$		<u> &gt;&lt;</u>	><	<u> </u>	>~<	\\ \\  \		
P5.1 Obs	24.1	29.1	33 - 1	100.2	24.0	29.	133.5	45.G		
F50	55 . 13	60.42	54 -69	12.27	55.64	1/20 .42	65.1	73.1		
E.7.R.	1.342	2.019	2.16.2	1-915	1.839	2.019	2.176	7.414		
DEUTA P Coc	111.3	49.0	55.11	61.2	41.2	49.8	55.8	16 00		
VAIC	210.0	2 <u>45.7</u>	337.5	12.55 • %	1.01.5	225.3	₹38.2	१२८५.		
TS Our	947	907	195	1039	931	297	950	1041		
750 77°	463	931	977	10/15	457	323	977	1069		
micos !	1772	1%% 7	1935	12175	17760	1131	1987	2179		
Thes (A)	1737	1947	1049	3170	1719	1937	1941	રાયુ		
START TIME	19706	TIM:		TO	TALL RUY	Tiv	hrs.	min.		

IDIE TIME 3- hours

Gill PROUBL MILLIAMY POR THE 2011/1

TF-41 PERFORMANCE CALCUI	LATION RECO	)RD			14593	T5 CF TYPE			
ENGINE SERIAL NUMBER (	42563	SEQUE	NCE NUM	BER TO		DATE ( 4	5 Aug	35	
TEST CELL NUMBER	7	OPERATOR Salucher INSPECTOR Benzis							
	DATA	TO BE	CRED/RI	CORDED					
DATA	5 inch	سر ددری	12.71.0	501/1	611.0	60012	7	6-181.C	
DEW POINT	55	55	55	55	56	56	56	56	
PAMB	29.51	29.51	29.31	29.51	29.52	89.52	29.52	29.52	
Pl	29.35	29.32	29.30	29.78	29.36	29.33	24.30	29."8	
тı	73.	73•	72.	72.	71.	71.	21.	70.	
TF (Fuel Temp)	81.	54.	91.	92.	90.	92.	92.	93.	
. AL ARFA			45.	772	<u>-</u>				
LAB SG	.822	2 6	<u>っ。</u>		.82	31 @	0		
P3	379	432	479	532	384	434	482	562	
т3	749	795	836	884	748	791	833	907	
RES. VALUE			$\checkmark$	\ <u>\</u>	4.2597	4.2597	4.2597	4.759	
J BCX TEMP (JBT)			<i>/</i> •	•	_	95.	97.	100.	
FNO Obs	4510	10150	11500	12940	8780	100%	11550	13650	
FNC	8743	10457	11876	13393	9023	10384	11930	14143	
FNT	3853	10565	1986	13504	9133	10495	12011	14254	
FNT 77°	1:76	15595	12016	13538	9156	15521	1207!	14290	
NLO Obs	7405	7810	8170	8520	7410	7800	8190	8755	
NLC	7292	7691	8053	8448	7316	7695	8000	8645	
ино Ора	11705	11964	12220	12520	11710	11935	12210	1255	
NHC	1			1	h	11775			
. WFO Obs		6315				6305		<del> </del>	
WFC	5435	6531	7597	8764	5521	6527	7614	4:70	
WFC 77°	=	===	7	李	丰	-2	-	2	
P5.1 Obs	24.0	27.1	35.3	37.8	24.7	29.1	33.4	101	
PSC	55.00	60.41				60.40	65.01	72.15	
E.P.R.	1.838	2.019					2.172	24411	
DELTA P Obs	- I	43.6				45.8	55.8	5.2	
WAIC	200	214.0	237.5		212.0		238.3		
T5 Obs	980	1052	1110	1165	857	916	968	1048	
<b>1</b> 50 77°	490	1963		1179	874		985	1060	
TLQSC	176.7	1881	1990		1774	1876	1984		
TLQS (A)	1759	1364	1967		1749	1850		212	
START TIME:	STOP	TIME:		TO	TAL RUN	TIME:	hrs.	min.	

F-41 PERFORMANCE CALCU	LATION RECO	ORD		FNC CK TSN TSO	1859	TYPE	TEST F/	; [2]	
ENGINE SERIAL NUMBER	14.56.2	SEQUE	ENCE NUMB		71		53		
	T 179263	OPERATOR Sul INSPECTOR BUSICES							
100. 00.00 110.1221			D BE STORED/RECORDED						
DATA	6.MW	6 M/V	אומבטאנו	7.712	6.17.2	6.11/0	600,2	8.4.8	
DEW POINT			-2	4	<del>                                     </del>		<del> </del>	<del> </del>	
	53	53	53	53	\$5	53	55	55	
PAMB		39.54			I	29.55	1		
P1			39.32	<b>.</b>	1	29.37	1	27.32	
Tl	68.	છ∙	9.		65.	66.	ري. ال	<u>65.</u>	
TF (Fuel Temp)	10.	42.	45.	196.	93.	95.	196.	97.	
AL ARFA		<u>.772</u>			1				
LAB SG		790	0		·87	100 6	<u>သိ</u>		
Р3	379	438	483	565	380	432	185	553	
т3	7∂3	776	816	893	730	דרר	826	883	
RES. VALUE	$\times$	$\geq$	$\sim$	$\geq$	4,227	4.22	4.22	4.23	
J BCX TEMP (JBT)	$\rightarrow$	$\times$	$\times$	$\times$	87.	90.	92.	15.	
FNO Obs	8520	10240	1520	13810	8550	10104	11550	1350	
FNC	क्ष	10553	11908	14315	8721	10397	11927	13975	
FNT	8873	10669	12019	1	1	10508	1		
FNT 77°	88~15	10696	12049		į.	10534		!	
NLO Obs	1319	794	9109			7770	8165	8155	
NLC	7284	7757	<b>४०</b> ी।	8714	7278			855	
NHO Obs	J 1	11588	18113				["	T	
NHC		11819		12525					
, WFO Obs	5178	6398	7299	l .		6250	1	1	
WFC ·	-5 379		7605		5403		7673	9253	
WFC 77°		X	X	X	Ž			1	
P5.1 Obs .	24.2	29.5	336	40.7	24.4	29.2	34.0	39.5	
P5C			65.03			60.47		1	
E.P.R.		2.044		2432	ı			Ţ.	
DELTA P.Obs		49.9	1	ا ما	41.7	1	56.5		
WAIC	210.4		288.9			225.5		1	
T5 Obs '		1024	1077		833		955		
T5C 77°	963	1063	1118	1230	313	925	1	1	
TLQSC	1753	1343			1	1	<u>१८२</u>	N59	
TLQS (A)	1692	1424	1115	90.19 31.19		1877	<del>                                     </del>	2140	
START TIME:		TIME:	11110		TAL RUN		hrs.	ZOB.	

F-41 PERFORMANCE CALCU	LATION RECO	ORD		FIIC CK TSN	143:13	TS CI TYPE	161 F/	R	
				TSO			PLATE 85	भे ।	
ENGINE SERIAL NUMBER		<del></del>	OPERATOR SAL INSPECTOR BOLISM						
TEST CELL NUMBER 2		OPER/		) VIT		I NS PECTO	RDOEIC	5171	
	SININ	TO BE S		CORDED	Selle	5.4/2	(serve)	71111	
DATA	1		3 797	4	5 2	5 14/2	600	71111	
DEW POINT	53	53_	53	53	53	53	53	53	
PAMB	29.51	29.21	31.21	2151	I	31.5)		21.20	
Pl	39.34			%.श्र	39.35	73-35	29:31	29.98	
Tl	64.	64.	64.	<u>63.</u>	63.	63.	<u>63.</u>	<b>√</b> 2•	
TF (Fuel Temp)	ਪ3.	94.	15.	55.	<u>92.</u>	9.	CH.	น.	
AL ARFA	4	5.772			· 	· · · · · · · · · · · · · · · · · · ·		<u>.                                    </u>	
LAB SG	• 83	39 @ 6	0 °		• 89	21 0 6	<u>)</u> •	·	
Р3	717	432	484	575	389	432	444	525	
т3	727	774	899	907	. 729	772	819	703	
RES. VALUE	4,000	4,000	4,000	4,000	4:157	4.137	4.137	4.137	
J BCX TEMP (JBT)	10-	89.	11.	<b>45.</b>	84.	88.	90.	92.	
FMO Obs	9500	10090	11600	14680.	865O	00101	11590	14000	
FNC	8745	10406	11972	14565	8899	10413	11930	1422	
FNT	8856	10517	19109_	14675	9010	16701	19011	14636	
FNT 77°	8878	10543	Į ·	14712	9032	10555	12121	14673	
NLO Obs	DIE	7740	3150	ક્કા૦	7355	7750	8150	8510	
NLC	7362	7689	5016	4770	7313	7706	शाय	4769	
NHO Obs	11555	11840	I	19655		11839	1		
NHC	บหาย	BAII	1200	12584		11762	19018	1253	
WFO Obs	5130	6310	7318	9295	5261	6035	7335	9340	
WFC	5343	5480	7642	4775	5483	Υ	7682	9823	
WFC 77°									
P5.1 Obs	23.1	28.9	37,-1	41.4	24.5	1-196	83.9	41.3	
PSC	54.94	i				60.45		73.4	
E.P.R.	ı	2.013	l	l .		8.090		2.45	
DELTA P Obs		48.6		68.0	41.8	1	55.0	68.2	
WAIC	210.0	8.166		1		624.7		259.	
T5 Obs	1-68	886	941	1035	835		948	1040	
т5с 77 <sup>0</sup>	<i>§</i> 53	919	975	1073	870	908	987	1801	
тьqsc	1751	1859	1956	3185	1769	18%	1997	2195	
TLQS (A)	1578	1813	1966	3117	1711	1:65	1128	3119	
START TIME:		TIM:	·····	<del></del>	TAL RUN		hrs.	min.	

F-41 PERFORMANCE CALC	JLATION REC	ORD		FAC CK TSN TSO	14393	TYPE	TEST F	3 P 899
ENGINE SERIAL NUMBER	1075 -	SEQUE	NCE NUME	BER 7	3.1	DATE 16		<u>85</u>
TEST CELL NUMBER		OPER/		URTUN	<del></del>	INSPECTO		<u> </u>
	DATA	TO BE S				·	~~	<del></del>
DATA	Gain	4 M . W	6121	7,110	5	6	7	В
DEW POINT	55	55	55	55				
PAMB		29.56					•	
Pl		79.36						
Tl	1	67.5		ાક	•			
TF (Fuel Temp)				99.	•			
AL AREA		45,	772					
LAB SG	.27	1.170 6			•	<b>(3</b>	0	
P3	383	435	485	550				
Т3	739	785	830	893				
RES. VALUE		4,076		4.076				
J BOX TEMP (JBT)				77.		•		
FilO Obs	872	10100	11600	13400				
FNC	8935	10398	11970	13864				
FNT		10507	l					
FNT 77°		10535			-			
NLO Obs	1	7795						
NLC	7314	וודר	8034	35 <i>3-1</i>				
ино Орв	11655	11900	I					
NHC	11543	סטרוו	12067	12432				
, WFO Obs		63 20	i					
WFC	5492	6546	7681	9250				
WFC 77°								
P5.1 Obs	24.7	29.3	34.1.	39.5	•			
P5C	55.72	60.CO	65.69	71.40	•			
E.P.R.		2.025						
DELTA P Obs		49.3	I		•	•	•	•
WAIC		2261						
T5 Obs	7	903						
T5C 77°		928	982	ルじらっ				
TLQSC	· 1	1879		7/45				
TLQS (A)		1838		アンナララ				
START TIME:		TIME:			TAL RUN	TIME:	hrs.	min.

TF-41 PERFORMANCE CALCULA	TION RECO	ORD		FIIC CX TSX TSO	14393	TYPE	N /Z/L			
ENGINE SERIAL NUMBER 14	2594	SEOUE	NCE NUME							
TEST CELL NUMBER			OPERATOR INSPECTOR							
			O BE STORED/RECORDED							
DATA		5/4/N	10mm 2	577	3191'V	5160	371170	8277 6!		
DEW POINT	49	49	49	49	48	4/9	48	43		
PAMB	29.90	29.90.	24.40	29.90	21 -90	19.40	<del> </del>	27.90		
Pl							29.64	7		
Tl		55.	56.	57.	62.	62.	62.	61.		
TF (Fuel Temp)	75.	28.	82.	80.	78.	81.	83.	841.		
, AL ARFA			45, 8	49						
LAB SG	18.	826	0	·	• 87	030 6	0 °			
Р3	375	425	475	565	373	420	470	590		
т3	703	741	78K	869	713	753	793	910		
RES. VALUE	$\searrow$	$\times$	$\geq$	<u> </u>	4.288	4.288	4.733	4.288		
J BOX TEMP (JBT)	×	$\geq$	$\times$	X	92.	94.	97.	406		
FNO Obs	8460	9710	11420	13800	8370	9640	1/200	14630		
FNC	8600	9891	11661	14136	8500	9817	11440	15022		
FNT	8711	1000 2	11772	14246	8611	9927	11551	15133		
FNT 77°	8733	10027	11801	14282	8632	7952	11585	15171		
NLO Obs	7190	7524	7941	8600	7190	7566	7949	2245		
NLC	1207	7541	7952	9603	7159	7533	7914	8815		
едо ОНИ	11456	11700	12040	12540	11490	11757	12071	12784		
NHC	11483	11727	12056	12545	11440	11706	12019	12741		
. WFO Obs	5129	5959	7083	8970	5075	5945	2015	9729		
WFC	5360	6231	7401	9402	5262	6171	7296	10177		
WFC 77°	X.		><	~			-			
P5.1 Obs	24.00	27.9	33.0	40.5	23.4	27.6	32 • 4	43.0		
P5C	54.76	58.64	64.19	72.08	54.12	58.50	163.62	74.83		
E.P.R.	1.829	1.966	2.144	2.41	1.808	1.956	2.126	2.501		
DELTA P Obs	10.8	47.0	55.0	66.8	40.2	46.8	54.2	70· S		
WAIC .	208.0	220.4	235.4	255.4	206.7	220.1	234.1	24.2		
T5 Obs	921	990	1047	1154	320	374	927	1054		
T5c 77°	975	1036	1103	1211	855	911	965	1096		
TLQSC	1765	1839	1966	2145	1746	1837	1956	2231		
TLQS (A)	1674	1745	1.65	1	1684	בררו	1889	2151		
START TIME:	STOP	TIME:		70	TAL RUN	ተ ነሳይ:	hrs.	min.		

IDIE TIME- INOUR 15 MIN AWITIONAL MILITARY TIME 25/11/

-41 PERFORMANCE CALCU	LATION RECO	RD		FNC CK	2264	TYPE	TYPE TEST		
		<del></del>	NOD 175	150	<del></del>	DATA PLATE			
HIGHE SERIAL NUMBER		· <del>`</del>							
EST CELL NUMBER	2	OPERATOR CARRENTUR INSPECTOR RULES,  A TO BE STORED/RECORDED							
	DATA 7 Mighies					<del></del> -			
DA TA			3	4	5	6	7	В	
DEW POINT	46	46	41	46					
PAMB	30.06	30 • 06	30.06	30.06	•			•	
Pl	29 -90		29.84	29.82				<u> </u>	
Tl	52.	52.	56					•	
TF (Fuel Temp)	61.	67.		75.	_ <u>•</u> l	<u> </u>		•	
AL ARFA			45.	829				·	
LAB SG		19 @ 6		r	·	<u> </u>	<del></del>		
Р3	383		480						
т3	705	744	790	854					
RES. VALUE				$\searrow$	İ				
J BOX TEMP (JBT)	<u> </u>	<u> </u>		<u> </u>	•	•	•	•	
FNO Obs	8540	10050	11570	13500					
FNC	5672	10182	11757	13756					
FNT	8743	10292	11867	13867		· .			
FNT 77°	4765	10318	(1897	13902					
NLO Obs	7231	7513	7768	8466					
NLC	7270	7614	1011	8512					
NHO Obs	11493	11695	12012	17417					
NHC	11556	11759	12 077	12485					
. WFO Obs	5271	6186	7318	4828					
WFC	5557	6516	7711	9314					
WFC 77°	>	$\geq$		>	$\times$	>	$\supset$	$\bigwedge$	
P5.1 Obs	24.4	29.9	33.6	39.8				•	
r50	55.00	59.68	64.64	71.13	•	•	•	•	
E.P.R.		1	7	2.377	1	•			
DELTA P Obs		F	56.1	1		•			
WAIC				252.9					
T5 Obs	9//	T	1037						
T5C 77°	772		1104						
TLQSC				2176			1		
TLQS (A)			1912						
ATART TIME:		TIME:			TAL RUN	TIME:	hrs.	min.	

I DLC TIME - 50 MWares

MAS LEMOGRE (40) 13700/14 (REV. 12-77) JEHONTI Meditional inclinity Time Beiner

F-41 PERFORMANCE CALC	JLATION RECO	RD			1 <i>५ । ५०</i> १५ <i>।</i>	TYPE TEST //			
ENGINE SERIAL NUMBER	142559	SEQUE	ENCE NUM		92				
	1200	OPER/		and		DATE /	R.R.	<u>- 1</u>	
<u> </u>		TO BE	STCRED/RI				.,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,		
DATA	SALINAIO	7miduin 2	7,111,2450	10 4/11/6	5	6	7	В	
DEM POINT	55	55	55	55					
PA MB	2960	29.60	2960	2960	•				
P1	29.44	29.41	29.38	29.36	•				
Tl	57.	£.	66.	65.					
TF (Fuel Temp)	90.	92.	94.	95.	•				
AL AREA			45	.78	8				
LAB SG	. 82	35 e 60			•	œ .	0		
Р3	377	430	480	560					
Т3	732	778	825	901					
RES. VALUE	4,6488	4.64 68	4.6128	4.6481					
J BCX TEMP (JBT)	75.	.17.	101.	109.	•	•			
FNO Obs	8'440	10000	11440	13750					
FNC			11796						
FNT	£761	10389	11900	14329					
FNT 77°	£7 <b>8</b> 3			14365					
NLO Obs	7338	7730	8123	8695					
NLC	7267		8053	8628					
рно Ора				12685					
NHC	11566	11817	12090	12587					
WFO Obs			7313						
WFC	5403	6494	7610	9585					
WFC 77°								-	
P5.1 Obs	24.0	29.0	33.5	40.6					
PSC	54.93	60.23	65.02	72.58	•				
E.P.R.	1.836	7,013	2.173	2425	•	•		L.	
DELTA P Obs	40.4	47.8	55.0	66.6	•				
WAIC	20%	222.9	256.5	255.2	•				
T5 Obs	842	900	958	1044					
TSC 77°	557	928'	955	1075					
тьqsc	1765	1881		2190					
ThQS (A)	1723		1	2131					
START TIME:	STOP	TIME:		τσ	TAL RUN	ተገላዊ:	hrs.	min.	

APPENDIX B: FORTRAN Program Documentation and Source Coding

...C-\* PROGRAM TF41

C-\* PURPOSE:

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THIS PROGRAM CALCULATES A CORRELATION COEFFICIENT TO ESTIMATE THE AMOUNT OF OXIDES OF NITROGEN PRODUCED DURING TF41 ENGINE TESTING. THE CORRELATION COEFFICIENT IS EXPRESSED AS AN "EMISSIONS TO FUEL RATIO" WHICH ESTIMATES TOTAL NOX EMISSIONS ON THE BASIS OF TOTAL FUEL USE ALONE.

# C-\* PROGRAMMER:

VERONICA J. HOBAN AIRCRAFT ENVIRONMENTAL SUPPORT OFFICE NORTH ISLAND, CA AUGUST 12, 1987

## VARIABLES:

<b>-</b> *	IDENTIFIER	TYPE	DESCRIPTION
-*			
C-*	KOUNT	INTEGER	# OF ENGINE TEST RUNS
^-*	N	INTEGER	# OF ENGINE POWER SETTINGS PER TEST
-*	IESN	INTEGER	ENGINE SERIAL # (LIMITED TO 6 DIGITS)
<b>∵−</b> *	J,K,L	INTEGER	MONTH, DAY, YEAR OF ENGINE TEST RUN
C-*	RPM	REAL	ENGINE POWER SETTING IN rpm
: <b>-</b> *	THRUST	REAL	ENGINE THRUST IN POUNDS
J <b>-</b> ★	FUEL	REAL	ENGINE FUEL RATE PER POWER SETTING IN
C-*			POUNDS/HOUR
-*	TIME	REAL	TIME AT A GIVEN POWER SETTING IN MINUTES
- <b>*</b>	EI	REAL	EMISSION INDEX EXPRESSED AS POUNDS OF
C-*			OXIDES OF NITROGEN PER 1000 POUNDS OF
C-*		•	FUEL
:-*	FUELUSE	REAL	POUNDS OF FUEL USED PER POWER SETTING
*-ك	NOX	REAL	POUNDS OF OXIDES OF NITROGEN RELEASED
C-*			PER POWER SETTING
' <b>-</b> *	TFUEL	REAL	TOTAL FUEL USE IN POUNDS FOR THE ENGINE
; <b>-</b> ★	•		TEST RUN
C-*	TNOX	REAL	TOTAL AMOUNT OF OXIDES OF NITROGEN
^-*			RELEASED FOR THE ENGINE TEST RUN
<b>-</b> *	COEFF	REAL	CORRELATION COEFFICIENT FOR EACH ENGINE
C-*			TEST RUN
C-*	ACOEFF	REAL	AVERAGE CORRELATION COEFFICIENT FOR ALL
-*			TEST RUNS IN THE INPUT DATA FILE

# ADDITIONAL VARIABLES WERE USED AS TEMPORARY STORAGE

### INPUT:

C-\*

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-C-\*

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!-\* !-\*

C-\*

THE INPUT DATA FILE IS "ENGINE.DAT", AND READING OF THE DATA IS LIST DIRECTED.

THE 1st RECORD CONTAINS "KOUNT" WHICH APPEARS ONLY ONCE IN THE DATA FILE.

THE NEXT N RECORDS CONTAIN "THRUST, RPM, FUEL, TIME" WHICH MUST BE EXPRESSED AS REAL VALUES (DECIMAL POINT INCLUDED). EACH SUBSEQUENT SET OF ENGINE RUN DATA IS ENTERED IN A

SIMILAR MANNER.

## OUTPUT:

THE OUTPUT DATA FILE IS THE STANDARD OUTPUT DEVICE (LINE

```
C-*
      METHOD:
         THE CALCULATION METHOD USED IN THIS PROGRAM IS DESCRIBED
         IN AESO REPORT No. 4-85 (JULY 1985) & AESO REPORT No. 10-87
         (NOVEMBER 1987).
 ·-- *
       RESTRICTIONS:
         THIS PROGRAM IS DESIGNED FOR THE TF41 ENGINE.
                                                         IT MAY BE USED
C-*
         FOR OTHER ENGINE TYPES AFTER SOME MINOR MODIFICATION.
             REVISE HEADER FORMAT TO REPRESENT ALTERNATE ENGINE TYPE
 -*
             REVISE THE LINE EI=a*EXP(b*THRUST) TO INCLUDE VALUES OF
-*
             a AND b WHICH REPRESENT THE ALTERNATE ENGINE TYPE
C-*
             REVISE FINAL FORMAT STATEMENT TO REPRESENT ALTERNATE
             ENGINE TYPE.
 -+
^-*
       PROGRAM MAIN
       REAL*4 NOX
       OPEN (50. FILE='ENGINE.DAT', STATUS='OLD')
       OPEN(6, FILE='LPT1', STATUS='OLD')
       READ(50, *) KOUNT
       TCOEFF=0.0
       DO 10 I=1, KOUNT, 1
         READ(50, *, END=999) N, IESN, J, K, L
         WRITE(6,9000) I, IESN, J, K, L
3000
         FORMAT('1',T10,'TABLE ',I2,'. Emission of oxides of nitrogen',
                    1X, 'from the testing of'/
     1
     2
                    T10, 'a TF41 engine at NAS Lemoore',
                    1X, '(Engine Serial Number: ', I6, ') '//
     3
     4
                    T10, 'Date: ', I2, '/', I2, '/', I4//
     5
                    T11, 'RPM', 4X, 'THRUST', 3X, 'FUEL FLOW', 3X, 'TIME',
                    3X, FUEL USE', 3X, EI', 3X, POUNDS NOX'/
     6
     7
                    T19,'(lb)',5X,'(lb/hr)',4X,'(MIN)',4X,'(lb)'/)
         TNOX=0.0
         TFUEL=0.0
         DO 20 M=1,N,1
           READ(50,*) THRUST, RPM, FUEL, TIME
           EI=2.02*EXP(1.76E-4*THRUST)
           FUELUSE=FUEL*(TIME/60.)
           NOX=EI*FUELUSE/1000.
           TNOX=TNOX + NOX
           TFUEL=TFUEL + FUELUSE
           IRPM=JFIX(RPM)
           ITHRUST=JFIX(THRUST)
           IFUEL=JFIX(FUEL)
           ITIME=JFIX(TIME)
           IF (M.EQ.17) THEN
              WRITE(6,9050) I
              FORMAT('1',T10,'TABLE ',I2,' (continued)'//
9050
                         T11, 'RPM', 4X, 'THRUST', 3X, 'FUEL FLOW', 3X, 'TIME',
     1
                         3X, FUEL USE', 3X, 'EI', 3X, 'POUNDS NOX'/
     2
                         T19,'(lb)',5X,'(lb/hr)',4X,'(MIN)',4X,'(lb)'/)
     3
              WRITE (6,9100) IRPM, ITHRUST, IFUEL, ITIME, FUELUSE, EI, NOX
           ELSE
              WRITE(6,9100) IRPM, ITHRUST, IFUEL, ITIME, FUELUSE, EI, NOX
              FORMAT(' ',T10,I5,3X,I5,5X,I5,6X,I3,4X,F7.1,3X,
9100
     1
                      F5.2,3X,F6.2/)
           ENDIF
20
          CONTINUE
```

PRINTER "LPT1"). THE OUTPUT IS NOT STORED ON DISK.

```
COEFF=TNOX/TFUEL
         TCOEFF=TCOEFF + COEFF
         WRITE(6,9200) TFUEL, TNOX, COEFF
200
         FORMAT('0',T10,'Pounds of fuel used in test',9X,F7.1//
         T10, 'Pounds of NOx per test', 32X, F6.2//
         T10, 'Pounds of NOx per pound of fuel used in test = ',F7.5//)
٠٥
       CONTINUE
       ACOEFF=TCOEFF/FLOAT(KOUNT)
       WRITE(6,9300) ACOEFF
       FORMAT('1',T10,'THE AVERAGE CORRELATION COEFFICIENT FOR THE'/
9300
                  T10, 'TF41 ENGINE IS ', F7.5/)
     1
- 39
       CONTINUE
       CLOSE(50,STATUS='KEEP')
       CLOSE(6)
       STOP
       END
```